

Challenges of tribology in Space Mechanisms

Lionel GAILLARD

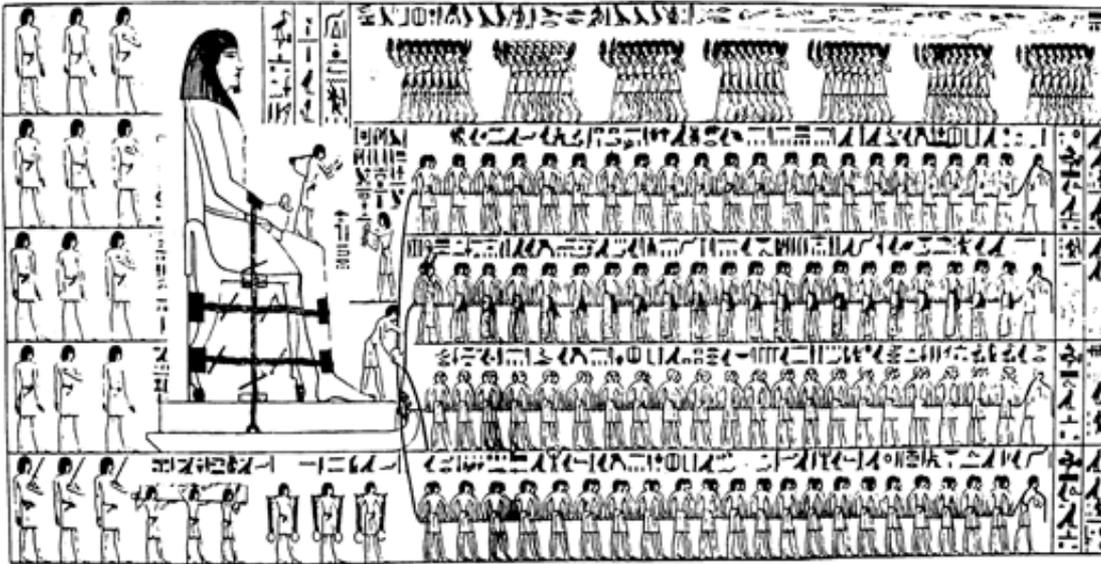
Head of the Mechanisms Section @ ESA/ESTEC

KIVI 03/11/2016

What is Tribology?

- Tribology = Tribos (to rub) + logos (study of)
- (*Jost Committee Report, a definition in 1966*) 'The science and technology of interactive surfaces in relative motion and related subjects and practices'
- But Tribology didn't start in 1966 - it's been around for ages...

Transporting an Egyptian Colossus c.1900 BC

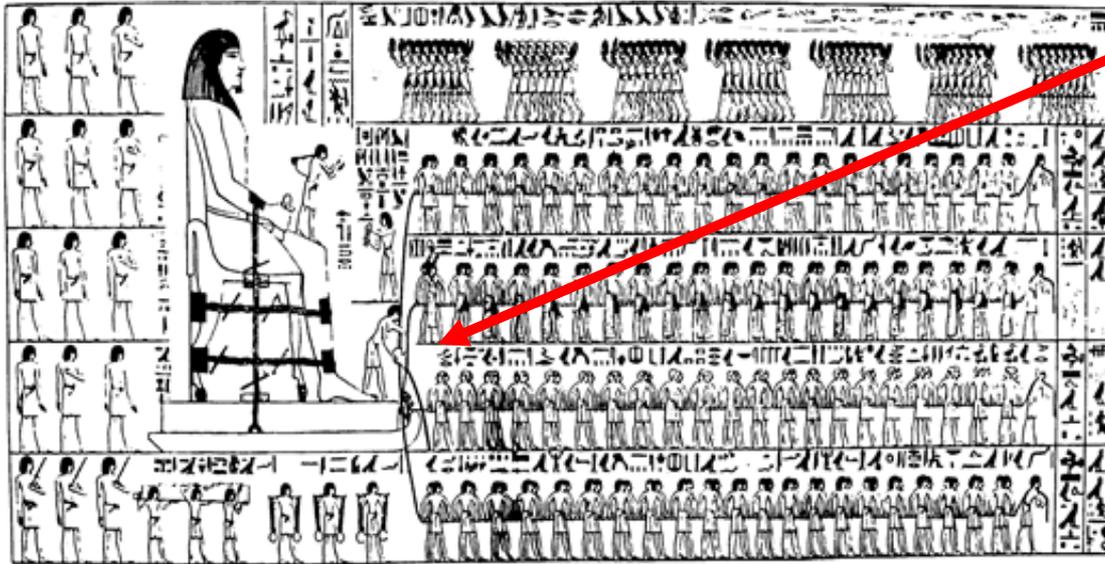


Colossus weighs $\sim 6 \times 10^5 \text{N}$
172 slaves
Each slave pulls with $\sim 800 \text{N}$
 $\mu = (172 \times 800) / 6 \times 10^5$
 $\mu = 0.23$

What is Tribology?

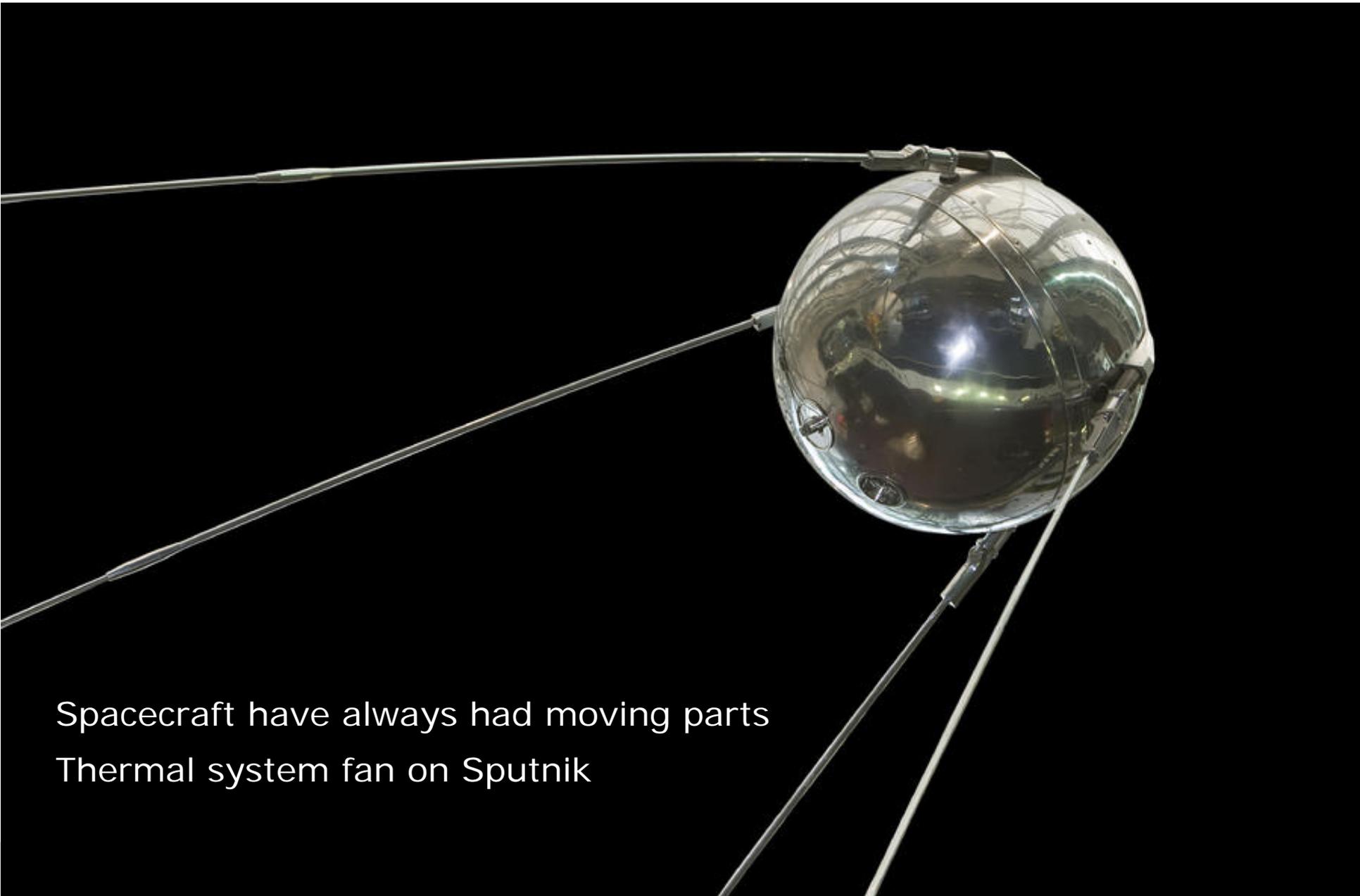
- Tribology = Tribos (to rub) + logos (study of)
- (*Jost Committee Report, a definition in 1966*) 'The science and technology of interactive surfaces in relative motion and related subjects and practices'
- But Tribology didn't start in 1966 - it's been around for ages...

Transporting an Egyptian Colossus c.1900 BC

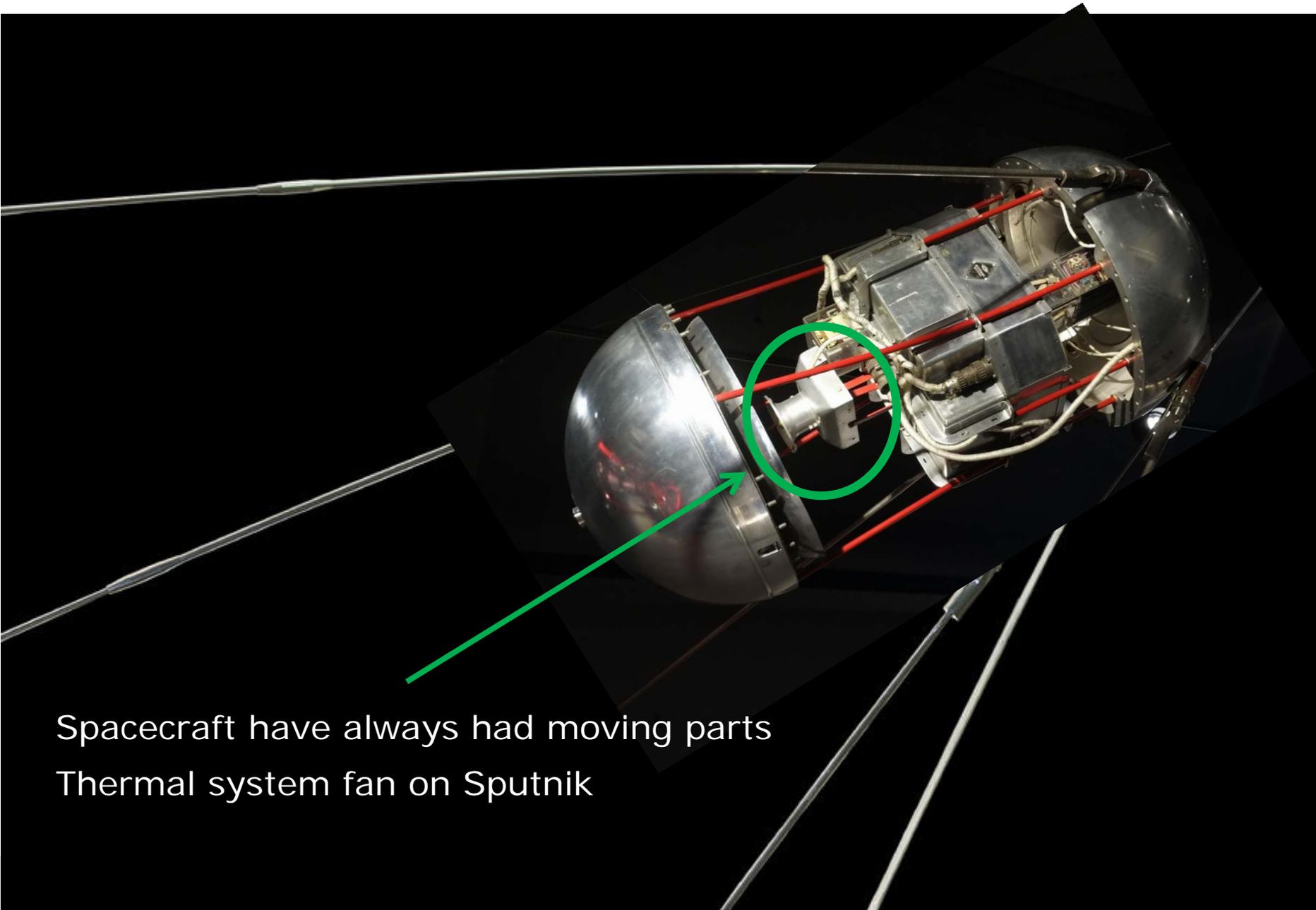


"Tribologist"
– applying lubricant

Colossus weighs $\sim 6 \times 10^5 \text{N}$
172 slaves
Each slave pulls with $\sim 800 \text{N}$
 $\mu = (172 \times 800) / 6 \times 10^5$
 $\mu = 0.23$

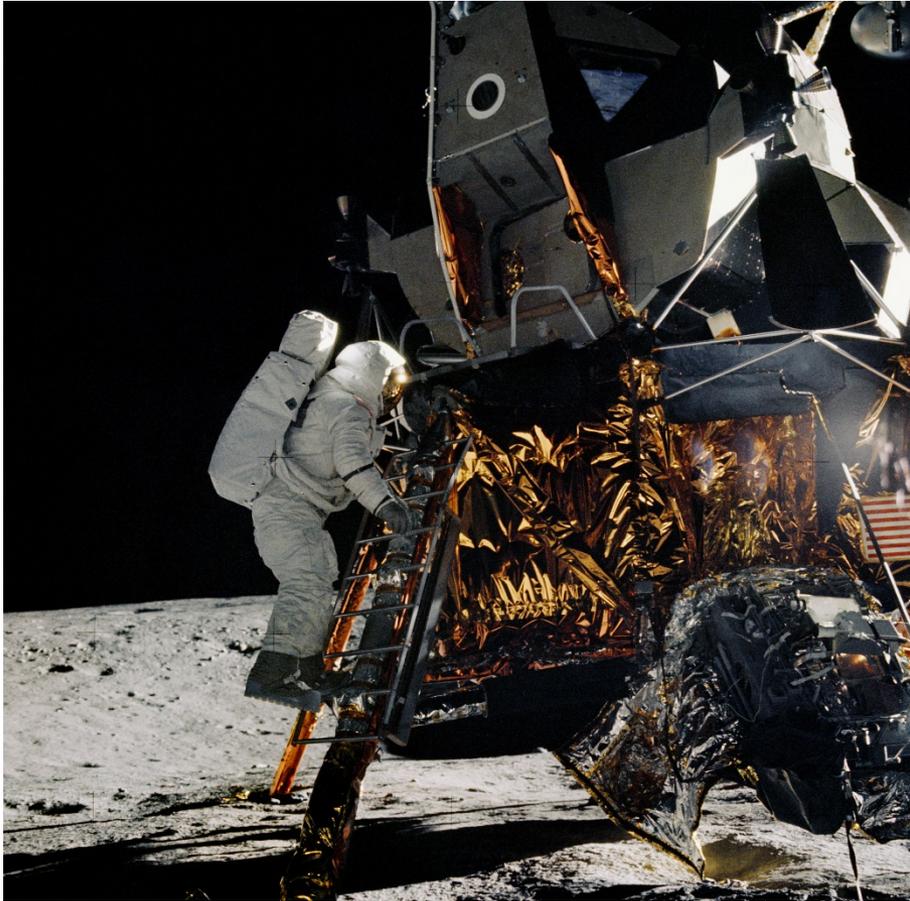


Spacecraft have always had moving parts
Thermal system fan on Sputnik

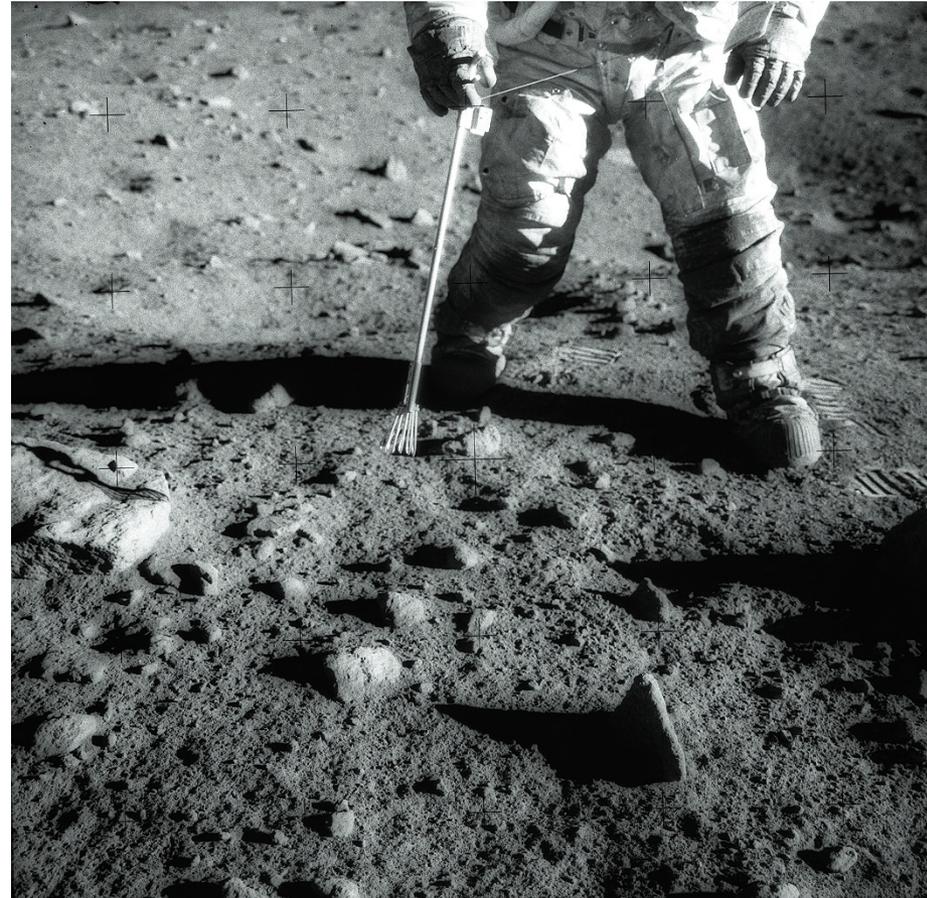


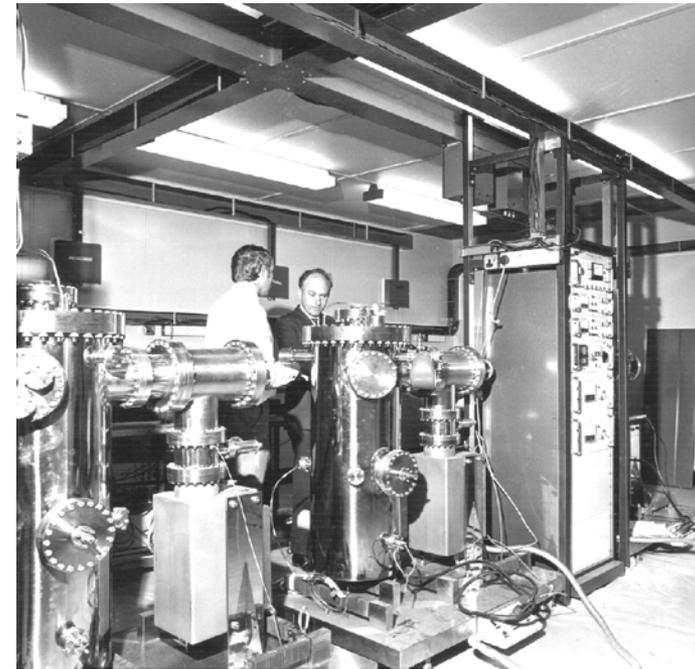
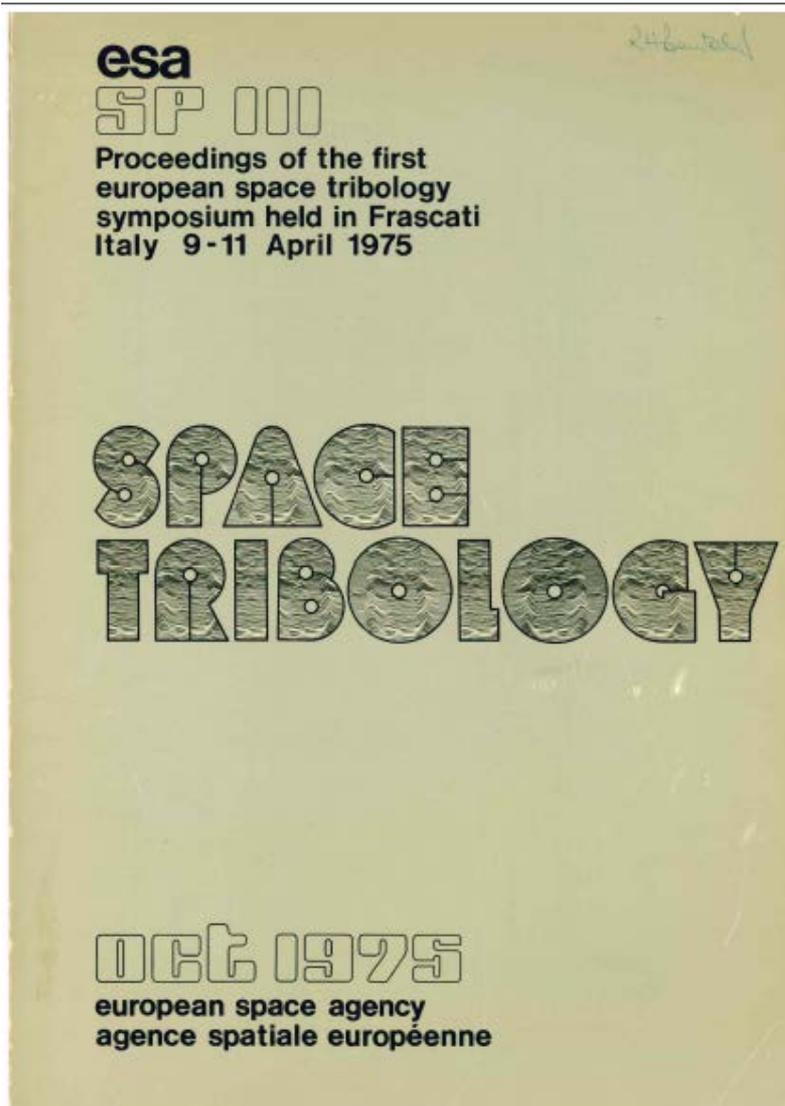
Spacecraft have always had moving parts
Thermal system fan on Sputnik

Apollo Program (1961-72)



- Moon dust highly abrasive
- Affected seals, optical surfaces and interlocking components



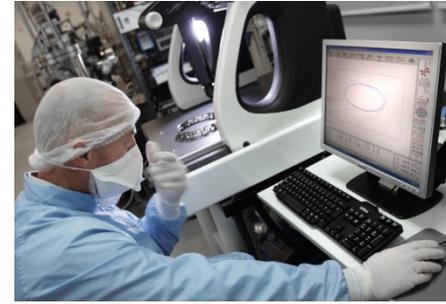


European Space Tribology Laboratory (ESTL)
Was established in 1972 and still active today

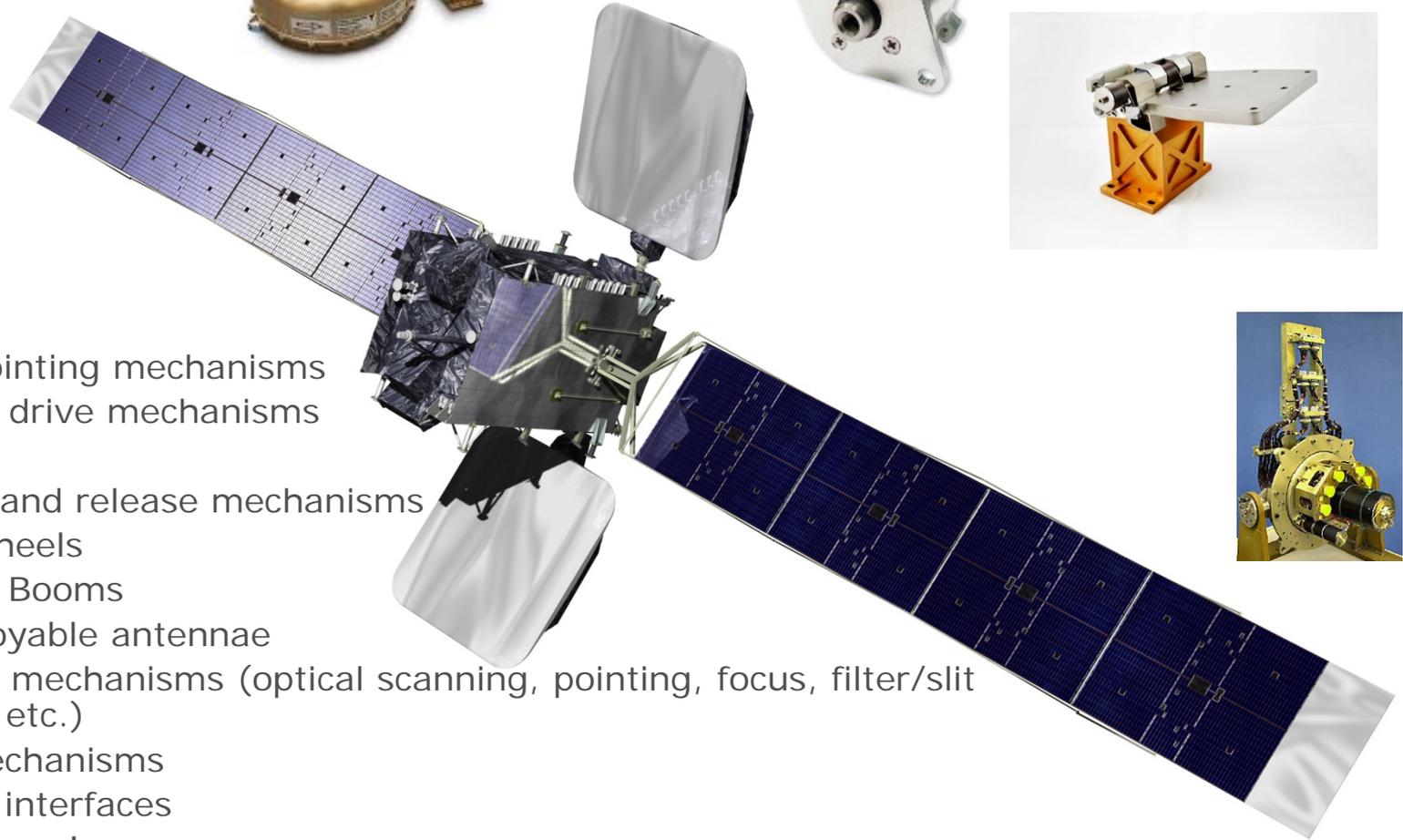
Fundamental objective :

*"to increase the efficiency and reliability
of spacecraft through the application of
good tribology"*

European Space Tribology Laboratory (ESTL)



Mechanisms on today Spacecraft



- Antenna pointing mechanisms
- Solar array drive mechanisms
- Hinges
- Hold down and release mechanisms
- Reaction wheels
- Deployable Booms
- Large deployable antennae
- Instrument mechanisms (optical scanning, pointing, focus, filter/slit positioning etc.)
- De-spin mechanisms
- Separation interfaces
- ...& many more!

A wide range of applications and life-times:



1 shot



~10,000,000,000 revs
(> 15 years)

All rely on a knowledge of friction and wear within the system and understanding of expected lifetime under the required operating conditions.

On Ground:

Manufacturing/Assembling (cleanroom environment)
Extensive testing (includes thermal vacuum and simulated launch)
Transportation (controlled temperature and humidity)
Storage (months to decades)
Integration at launch site (cleanroom)
Planetary protection (biologically clean for planetary missions)



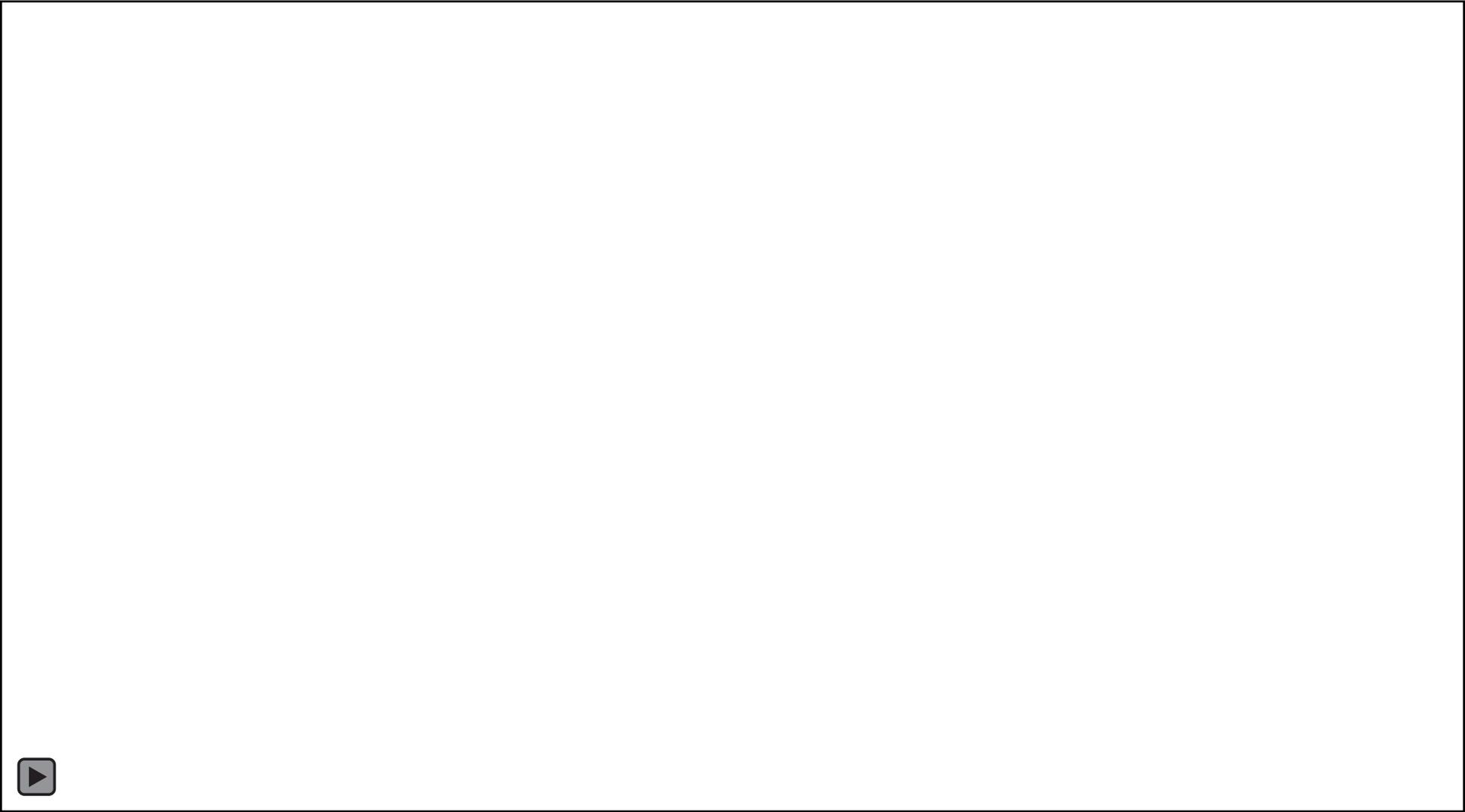
Then:

Launch (high levels of vibration and shocks, Pressure decrease)

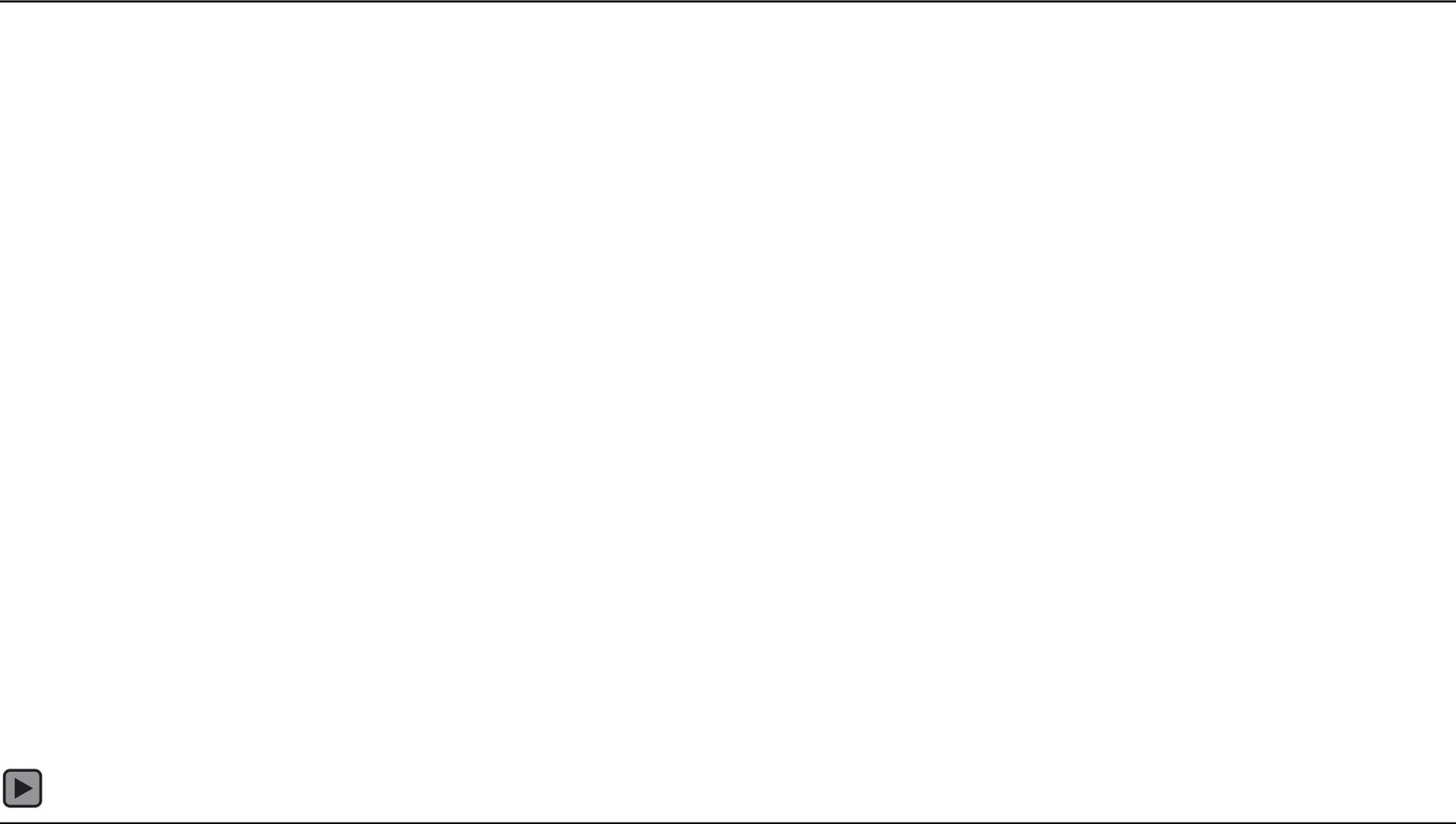
Finally:

Vacuum ($\sim 10^{-7}$ to 10^{-13} mbar)
Atomic Oxygen
Reactive environments (planetary missions may experience CO_2 , H_2 , traces of CH_4)
Temperature extremes (and high thermal gradients)
Radiation
Prolonged Stasis (long inactivity, followed by “must work” moment)
Micro-meteorite impacts ...

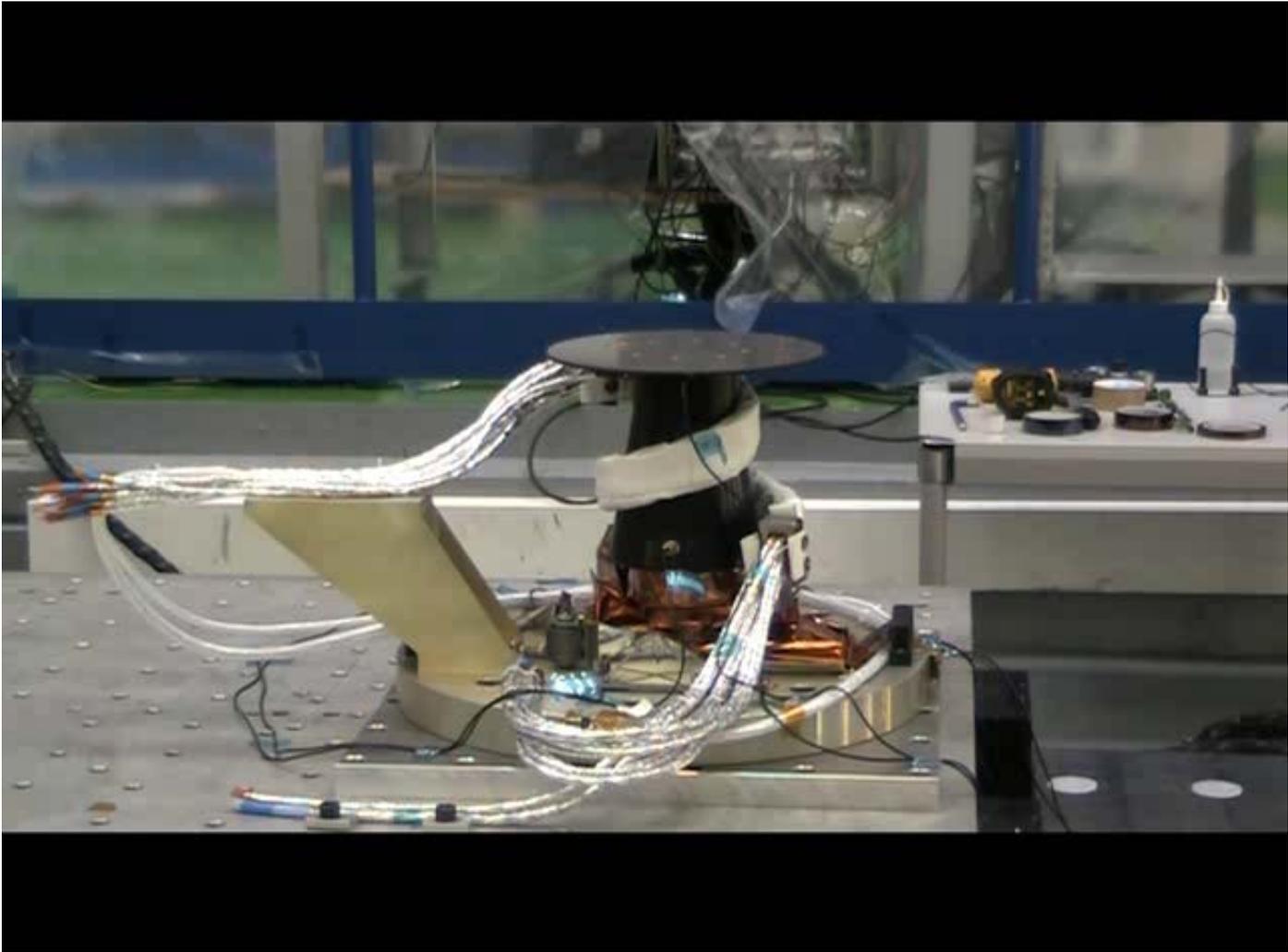
Example of mechanisms operation in space



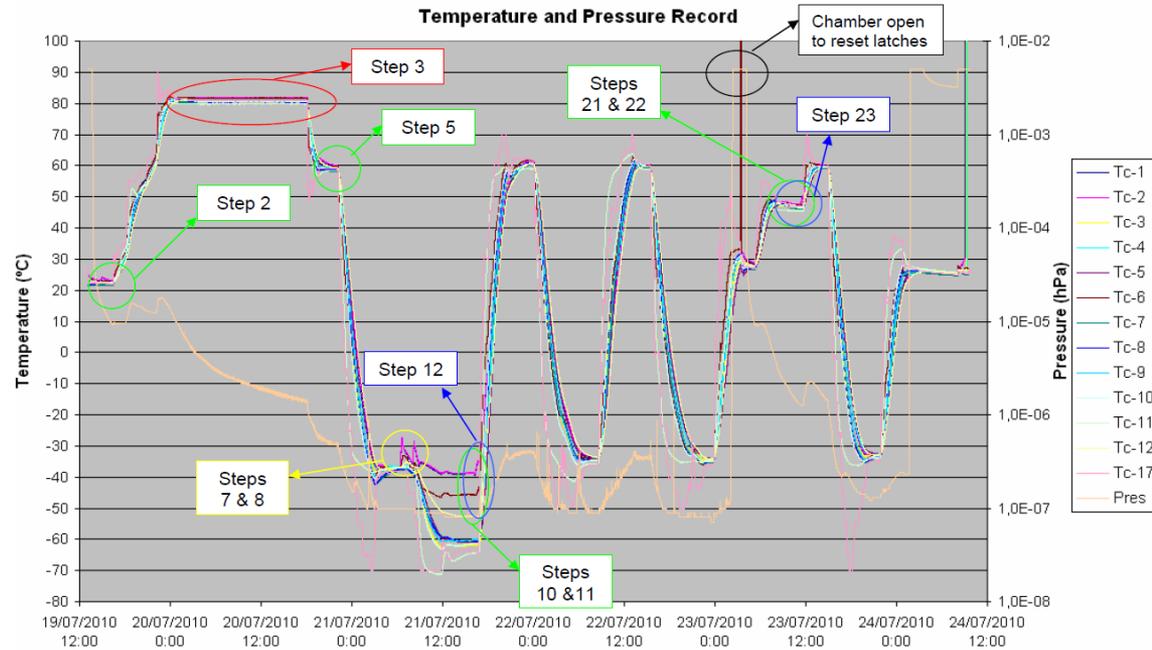
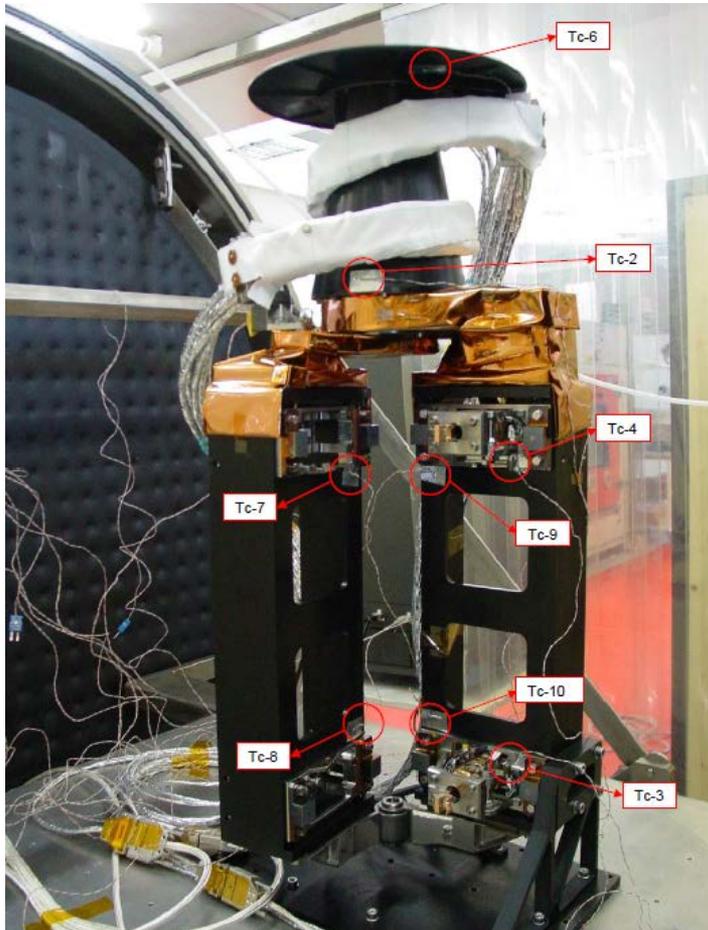
Mechanisms test on ground



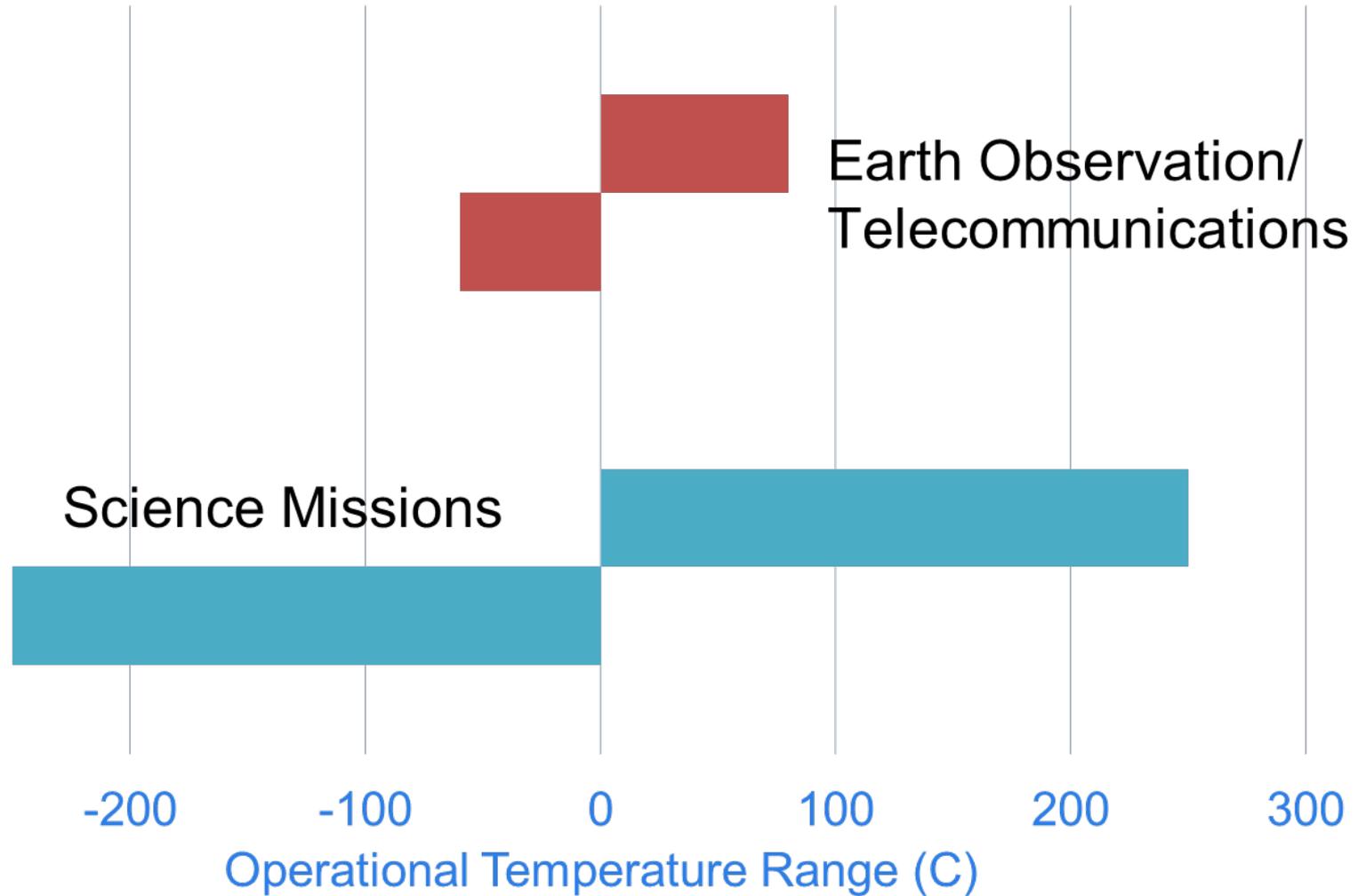
Vibration test on ground, example



Thermal Vacuum



Typical Temperature ranges





Definition:

'Apply a substance such as oil or grease to (an engine or component) so as to minimize friction and allow smooth movement'

(Oxford English Dictionary)

...Better Definition (permits solids):

'The separation or protection of surfaces in relative motion, in order to reduce friction and wear' or better control it (less variation)

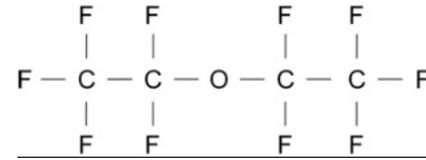
BUT

Conventional oils (and greases) would evaporate **within seconds** in the vacuum of space leaving a dry unlubricated mechanism and producing gross contamination of their surroundings.

So what CAN we use in space?

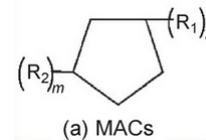
Multiply-Alkylated Cyclopentanes (MACs) or PerFluoroPolyEther (PFPE)

- Synthetic, long chain, chemically stable and inert oils, e.g. MACs and PFPEs (and greases derived from them, often thickened by PTFE).
- VERY low volatility
- Stability to shear and extreme pressures
- Typical Temperature
 - Minimum > ~ -60°C
 - Maximum ~ +80 to +100°C
- **BUT**
- Increasing viscosity when cold and evaporation limits high temperature



- Very very low volatility
- Long life in vacuum

part of a linear PFPE molecule in simplified form

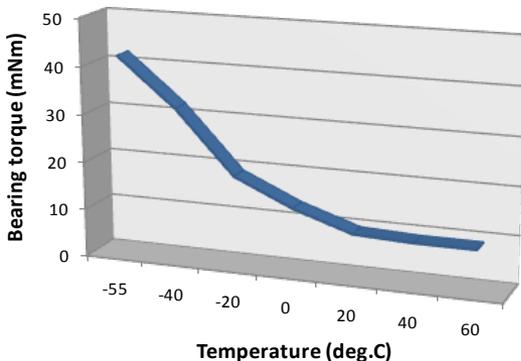


- Very low volatility
- Very Long life in vacuum

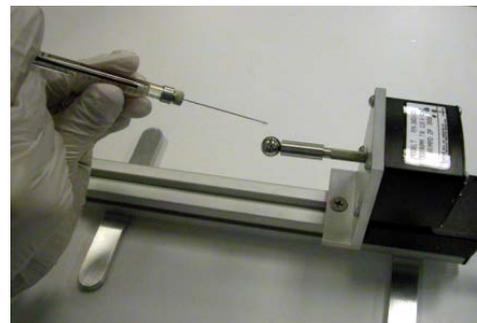
where R₁ and R₂ are hydrocarbonyl groups

Benefits:

- ❖ High thermal conductance
- ❖ Super smooth operation (low torque noise)



PFPE Oil (Z25)



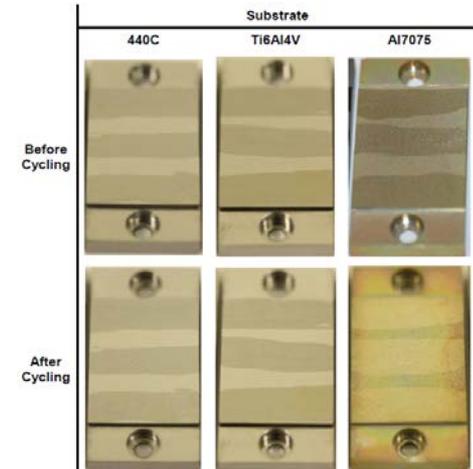
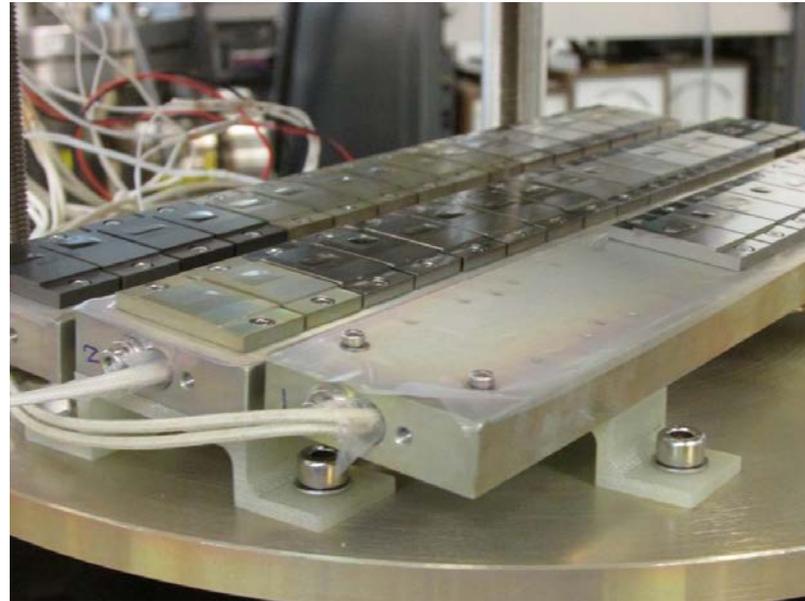
Anti-Creep Barriers are needed with Fluid Lubricant

Young's Equation

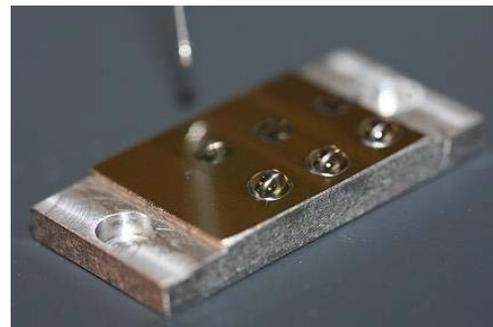
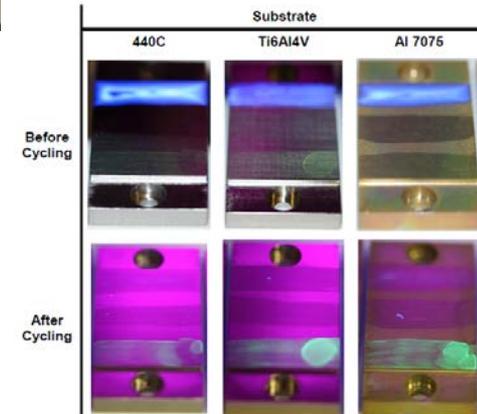
$$\gamma^{sv} = \gamma^{sl} + \gamma^{lv} \cos\theta$$

θ is the contact angle
 γ^{sl} is the solid/liquid interfacial free energy
 γ^{sv} is the solid surface free energy
 γ^{lv} is the liquid surface free energy

ramé-hart instrument co.



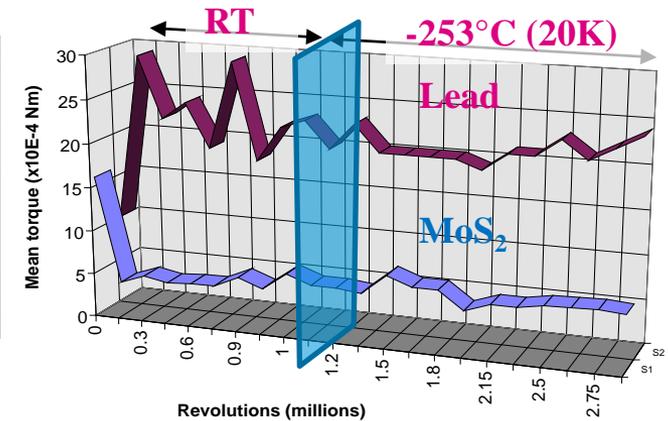
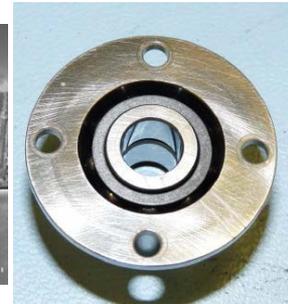
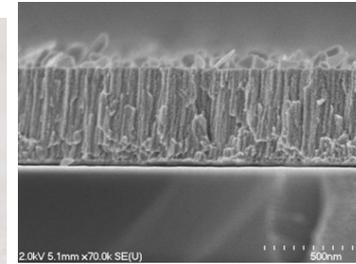
Under UV light



Solid Lubricants for Space

- Low shear strength metals (e.g. lead, silver)
- or lamellar solids (e.g. MoS_2 , WS_2).
- low-shear strength Polymer (PTFE, polyimide)

- Negligible volatility
- Can be insensitive to temperature
- Stable, but generally lower lifetime and more “noisy” than fluids.
- Permissible Temperatures
 - Minimum $> \sim -270^\circ\text{C}$
 - Maximum $\sim > +300^\circ\text{C}$
- **BUT**
 - Adhesion, thickness, life and replenishment may be issues
 - Operation in air may be forbidden or very severely restricted.



Benefits:

- ❖ Accelerated testing feasible
- ❖ Some are electrically conducting

Solid or Liquid Lubricant? That is the (first) question !

Conditions when only dry (solid) lubricants should be used:

- Temperature extreme (high or low)
- Contamination risk to other parts of spacecraft (evaporation or creep)

Conditions when only fluid lubricants should be used:

- Very long life required
- Significant running in air required (ground testing)

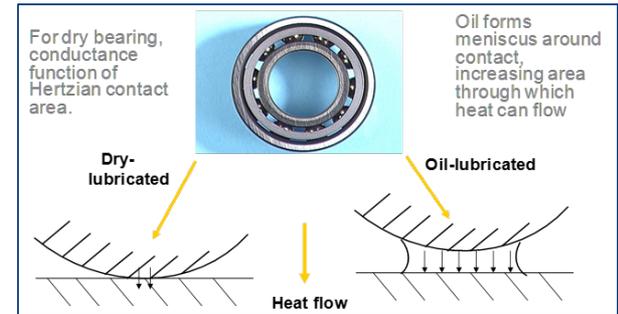
Other criteria:

- High thermal conductance
- Low torque noise
- Accelerated testing needed
- Low electrical resistance

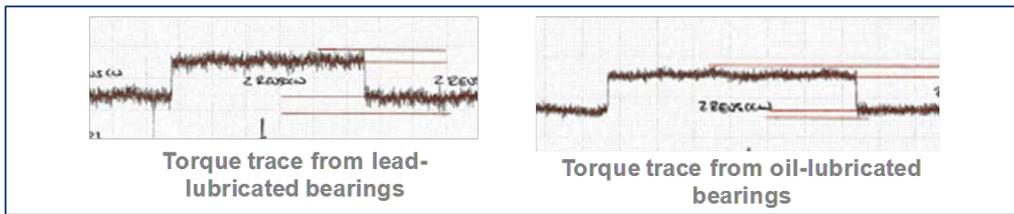


Fluid lubricant

Solid lubricant



Monolayer formation time:
 Oils - typical for best space oils:
 3 years at 0°C
 3 hrs at 100°C
 Lead ~30,000 years at 100°C



- Solution has to be effective in vacuum and at extreme temperatures
 - It has to work in space and on ground, if possible
 - Need to consider lubrication in early stages of design. (Impact on motors size; etc...)
 - Limited range of lubricants available (two base oils, then a couple of solid)
 - Re-lubrication in space is not normally an option
 - Qualification or acceptance of new lubricant solutions or other new surface treatments is a long process.
-
- **What flies may often be a compromise**
 - **the perfect “one fit all” lubricant for space does not exist...**
- therefore extensive testing and experience is needed for the lubricant selection and optimisation for each new application.**

- **Tribometer Level** : basic friction & wear for materials and coatings, thin film durability, relative performance of candidate lubricants in idealised but controlled conditions (Mainly Pin-on-Disc and spiral Orbit tribometer)
- **Component Level** : e.g. Ball bearing, gears, etc .. - assessment & validation of torque, life, efficiency, adhesion etc. of individual components. Gives increased confidence but may be difficult to “condition” test item in isolation – so how representative is the test?
- **Mechanism Level** : only valid way of qualifying/accepting a mechanism lubrication solution, but little information on individual component performance

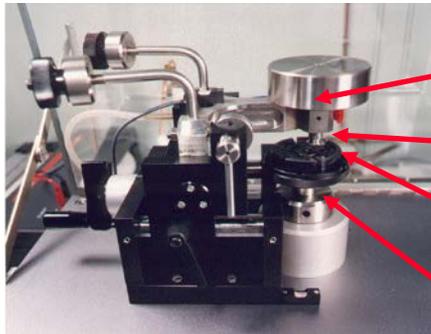
Pin-On-Disk Tribometer

Vacuum (2) and non-vacuum (2)
sliding tribometers

- Fully controllable (unidirectional / reciprocating)

Uses / recent studies:

- Investigation into the impact of surface treatment on the performance of thin solid lubricating films
- Characterisation of lubricant-free material pairings
- Development of improved lubricants

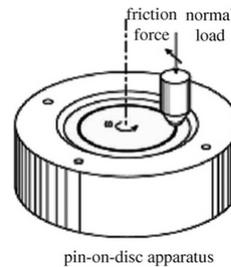


Deadweight load

Pin

Rotating disc

Drive-shaft



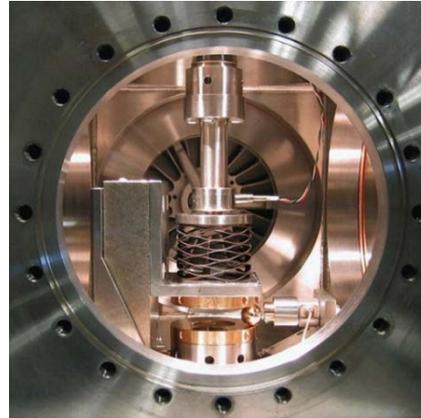
pin-on-disk apparatus

- Load: 10 - 25N
- Various pin diameters (stresses)
- Various pin/flat materials
- Speed: 0.1-500rpm
- Sliding Speed: $\sim 1e-4$ to 1m/s
- -100 to 200°C
- Cover gas to high vacuum

Spiral Orbit Tribometer

Unique facility – only machine outside of NASA

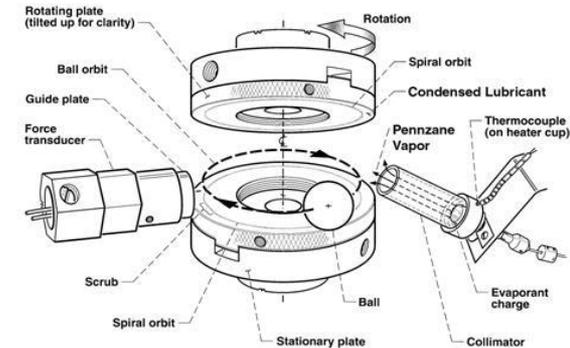
Rolling tribometer, more representative kinematics of a angular contact bearing



- Load: 10-225N
- Peak Hertz stress: 0.3 to 3GPa
- Ball diameter: 3.175 - 12.7 mm
- Speed: 1-200rpm
- Ambient to 100°C
- Various atmospheres including vacuum to $\sim 10^{-8}$ mbar

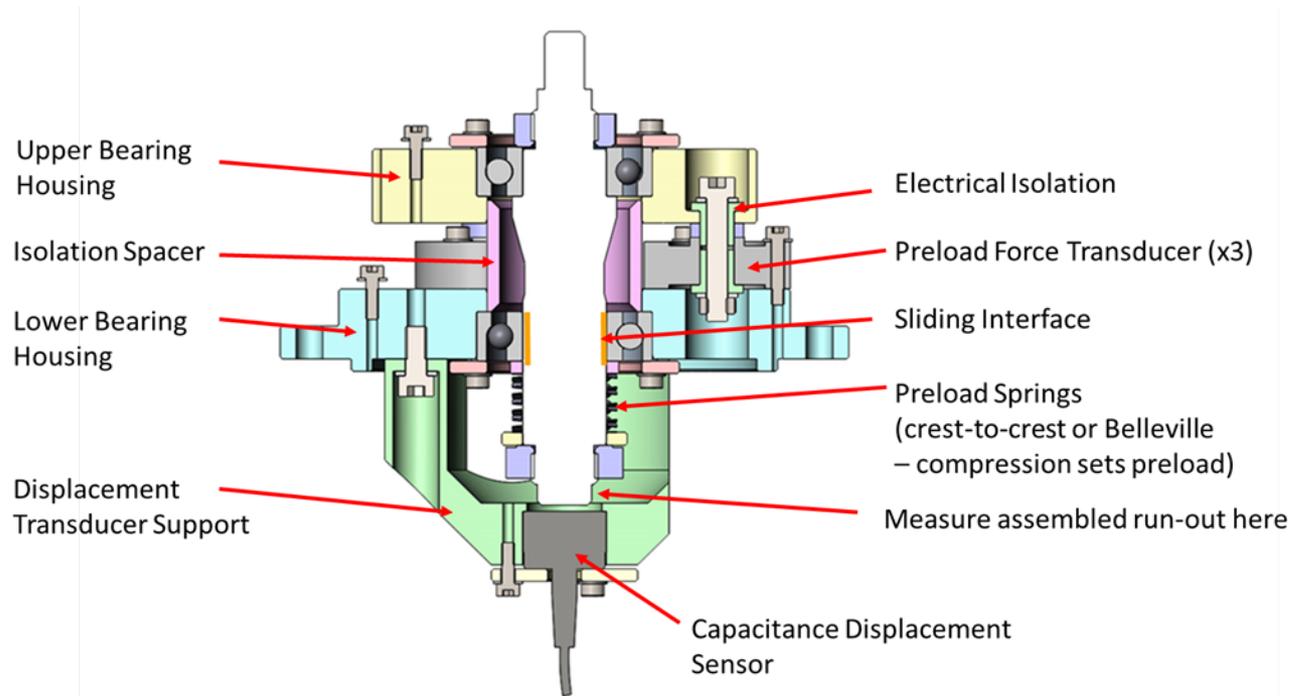
• Uses / recent studies:

- Evaluation and comparison of fluid lubrication (PFPE, MAC, ionic)
- In-situ analysis of fluid degradation behaviour (using mass spec.)
- Characterisation of lead replacements for thin film solid lubrication
- Study of transfer film lubrication, ACE etc.



Background:

Polder, EarthCARE, and some wheel programmes highlighted a lack understanding of fluid or transfer film health /formation at bearing level.



- Measured parameters

- Torque , axial shaft displacement (capacitive measurement of lubricant film thickness <10nm resolution), preload, film thickness (electrical resistance), temperature

Gear Testing – Unit Level (1/2)

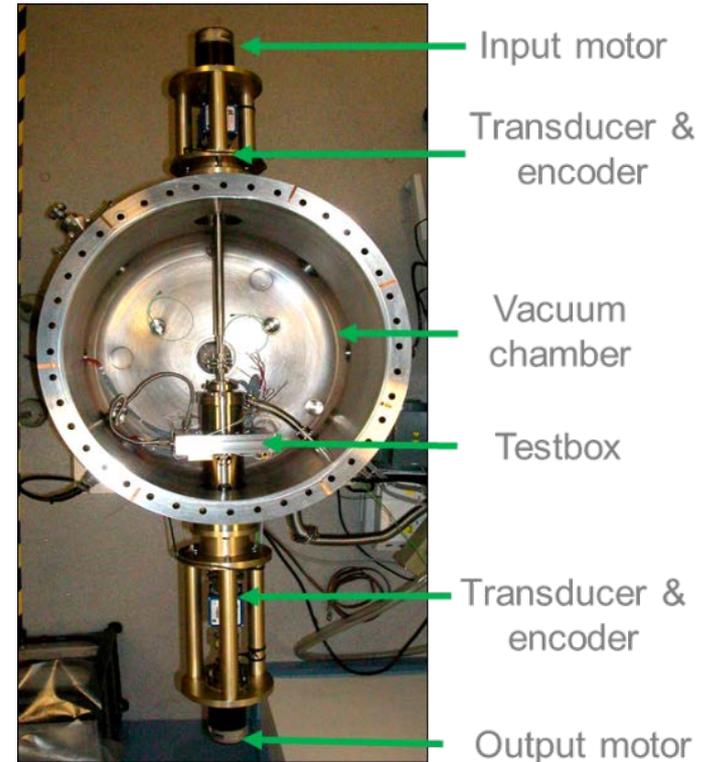
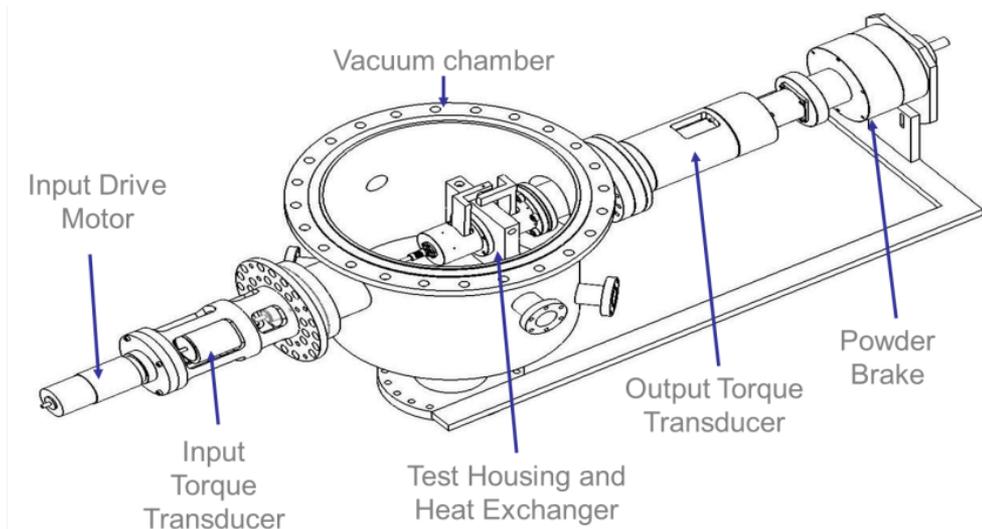
Torque to 400Nm (most recent tests around 4-30Nm)

Temperature: -40/+90°C (typ.)

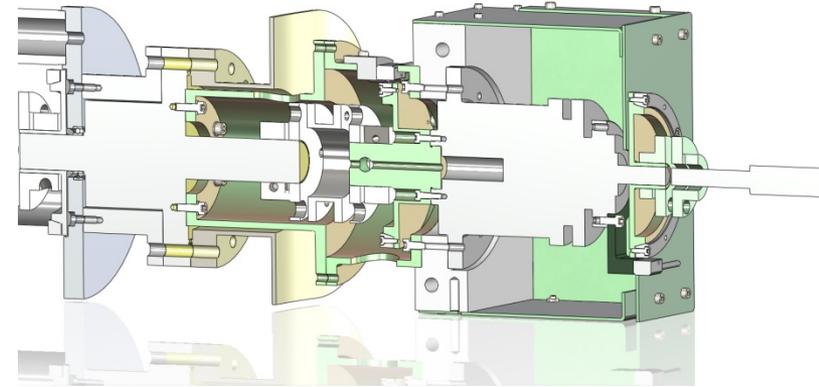
Speed: 0.1-500rpm at input (typically 50rpm)

Environment: Cover gas to high vacuum

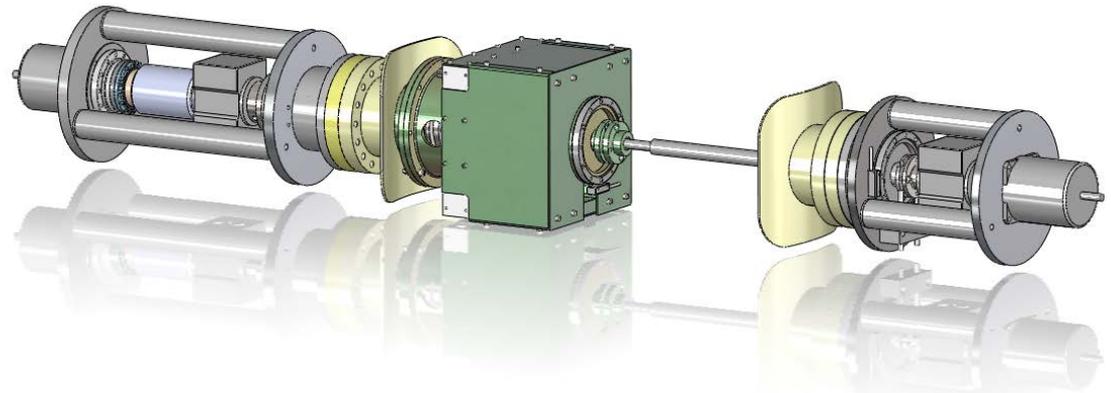
- Break-out torque , efficiency ,hysteresis, lost motion (backlash), torsional stiffness, no-load back-driving torque



- Programme Findings:
- Efficiency, speed and output torque behaviours PFPE/MAC
- Maplub PF100a and SH050a – flex-spline wave generator interface
- Life limit by oil LOSS from FS/WG interface
- Re-lubrication after run-in for maximum life (17 million i/p / 106250 o/p)
- ARTES 5.1 (HFUC7) – Maplub 100b/Nye 2001a - suspended



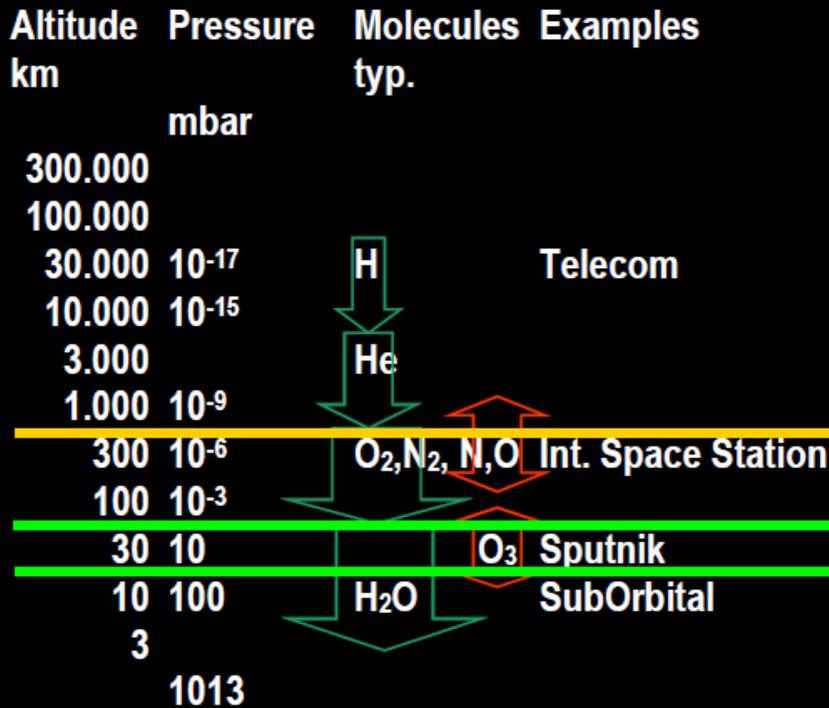
- 2015 HD in small angle oscillatory motion – proof of facility and lifestest (100k oscillations of $\pm 0.0025^\circ$)



- 2 Unbalanced magnetron sputtering (PVD) facilities (CF4 & CF5)
- Validated Space Processes for PVD – MoS₂, Lead, Silver
- Films typically 0.2-0.5µm on bearings, up to 1µm for other components
- Thinner films for industrial applications
- Future
 - Enhanced MoS₂
 - Lead Replacement
 - In-process RGA to support product development/quality monitoring
 - Higher and more automated bearing run-in capacity



Cold-Welding (also called galling ...)



Adhesion in
Contact type:
Static
Impact

Adhesion in
Contact type:
Fretting
→ Material !?

No adhesion in ambient
Presence of oxygen !!
Fretting corrosion !!

Vacuum level

High vacuum

- | Less than 10^{-8} mbar
- | Contamination time 100 sec

Low vacuum

- | 10^{-3} to 10^{-6} mbar
- | Contamination time < 1sec

Air (oxygen)

- | Pressure > 0,1 mbar
- | Contamination time < 0,00001 sec

Effect on surface

- Surfaces will not be re-covered
- Destruction of surface layers result in clean surfaces
- Adhesion very likely !!!

- Surfaces may become clean, e.g. under fretting
- Adhesion under certain criteria possible, e.g. fretting !!

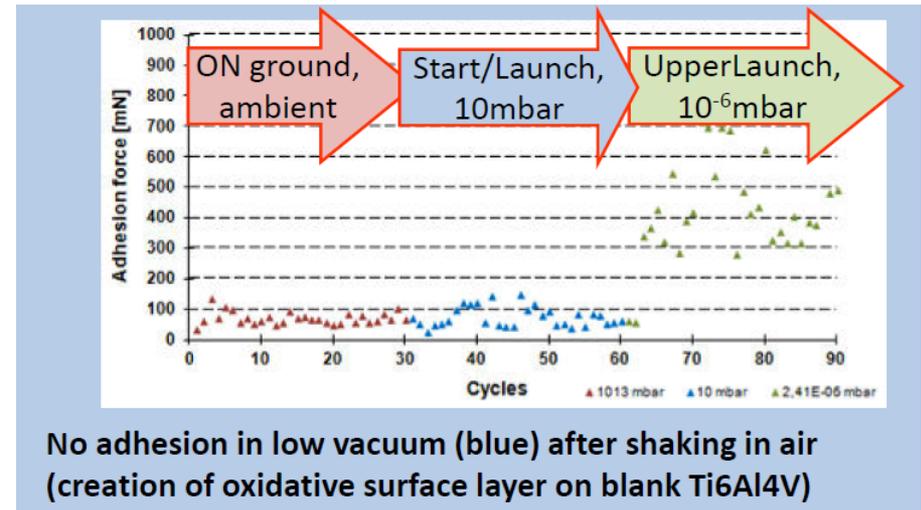
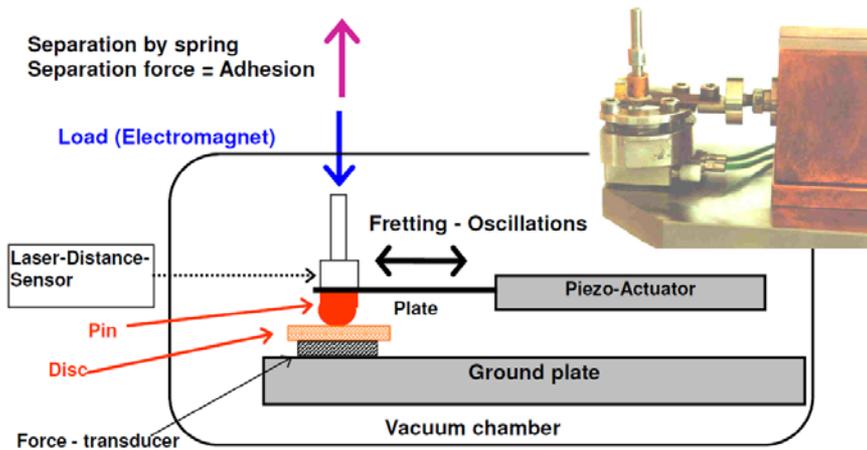
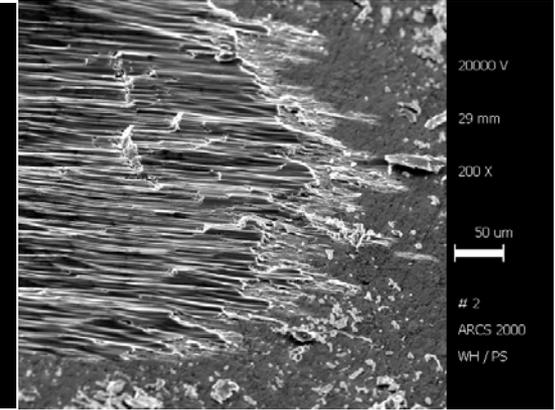
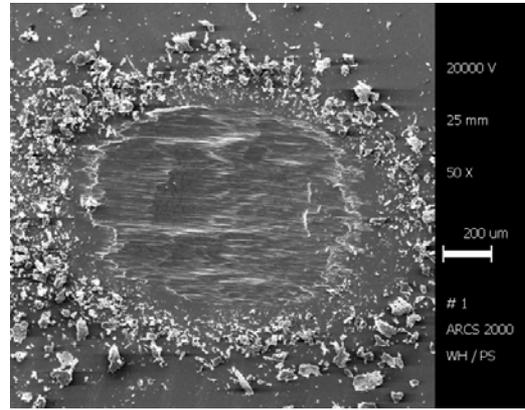
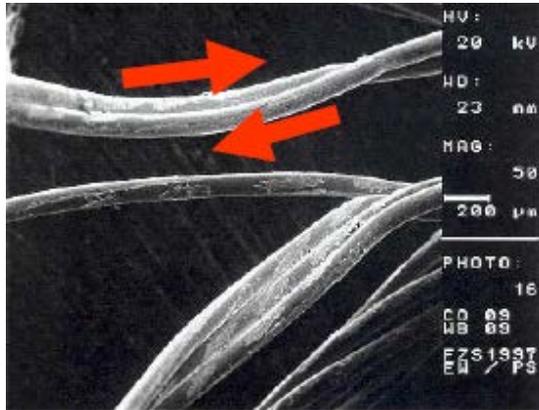
- Surfaces covered
- Metal-atoms of two bodies cannot come close enough
- No adhesion

Adhesion likely in contact types:

Static,
Impact,
Fretting

(Impact)
Fretting

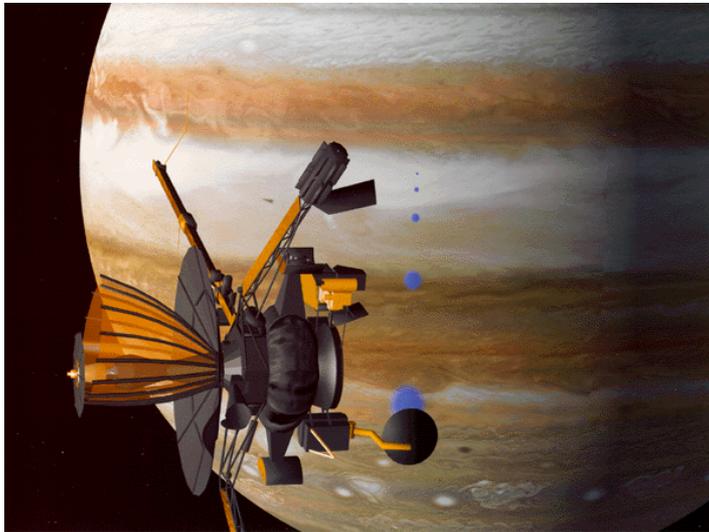
Exception:
Tightening of stainless steel screws



- Galileo Mission (US)



- Launched Oct 18th 1989 on STS34
- Jovian orbit insertion Dec. 1995
- \$1.4 billion mission cost

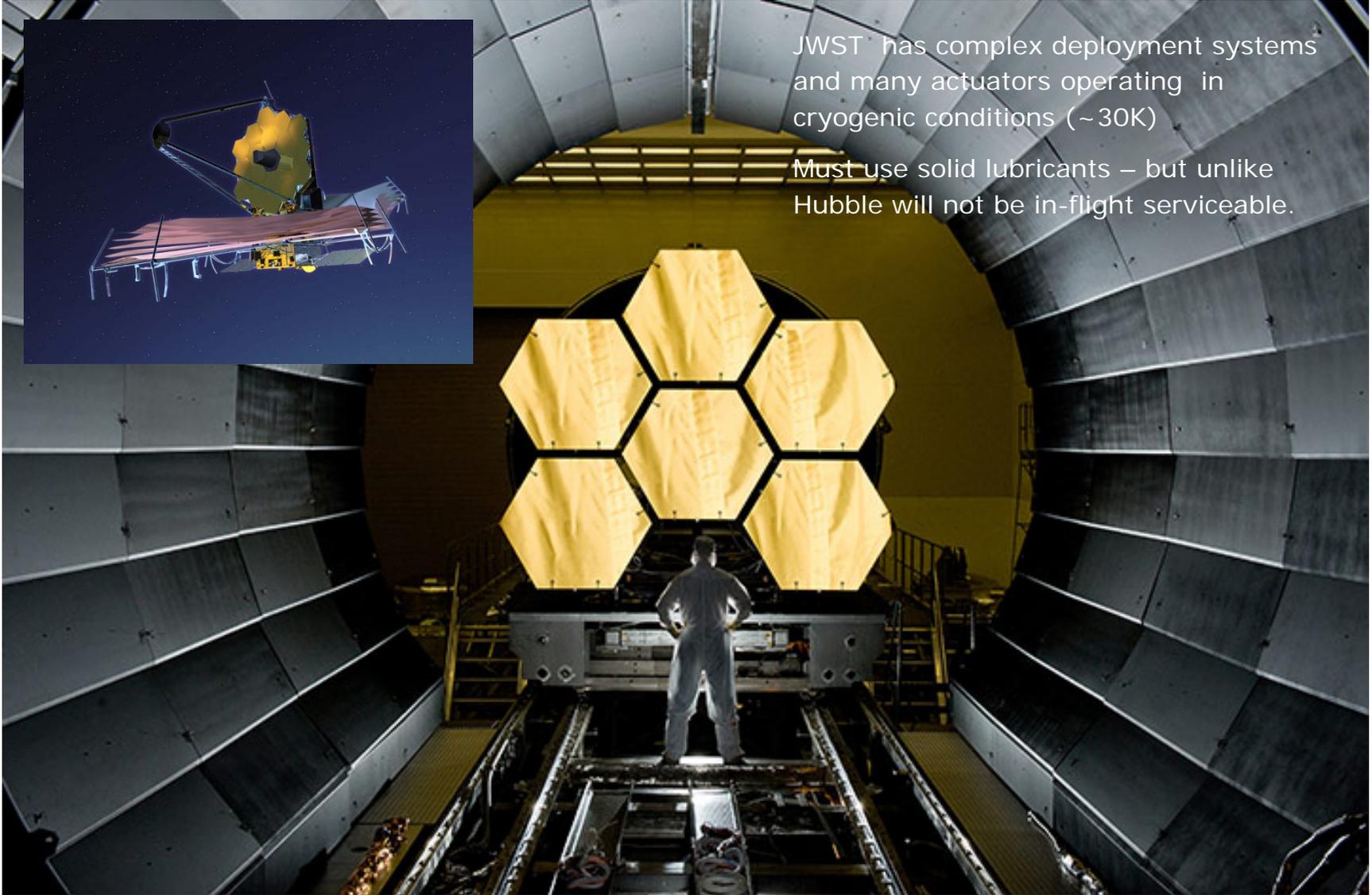


- High gain antenna failed to open
- Loss of 70% of its science data
- Reason:
 - ❖ MoS2 lubricant coating on antenna mesh wore away during long over-land journeys on ground.
 - ❖ Also spent 4.5 years in storage following Challenger disaster
 - ❖ Fretting and adhesion after transportation

Space Tribology knowledge Test



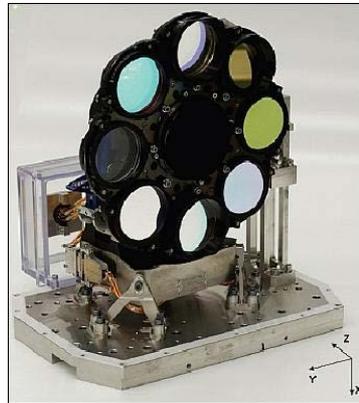
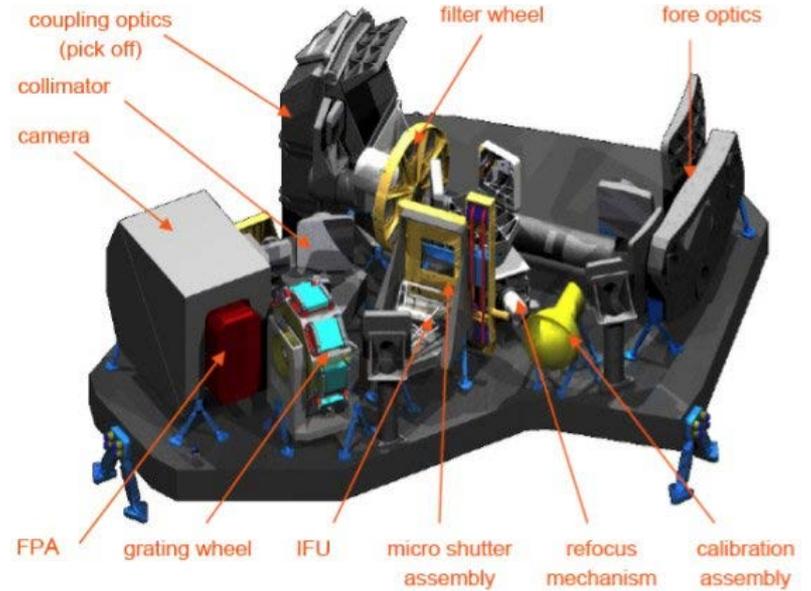
Solid Lubrication (30K)



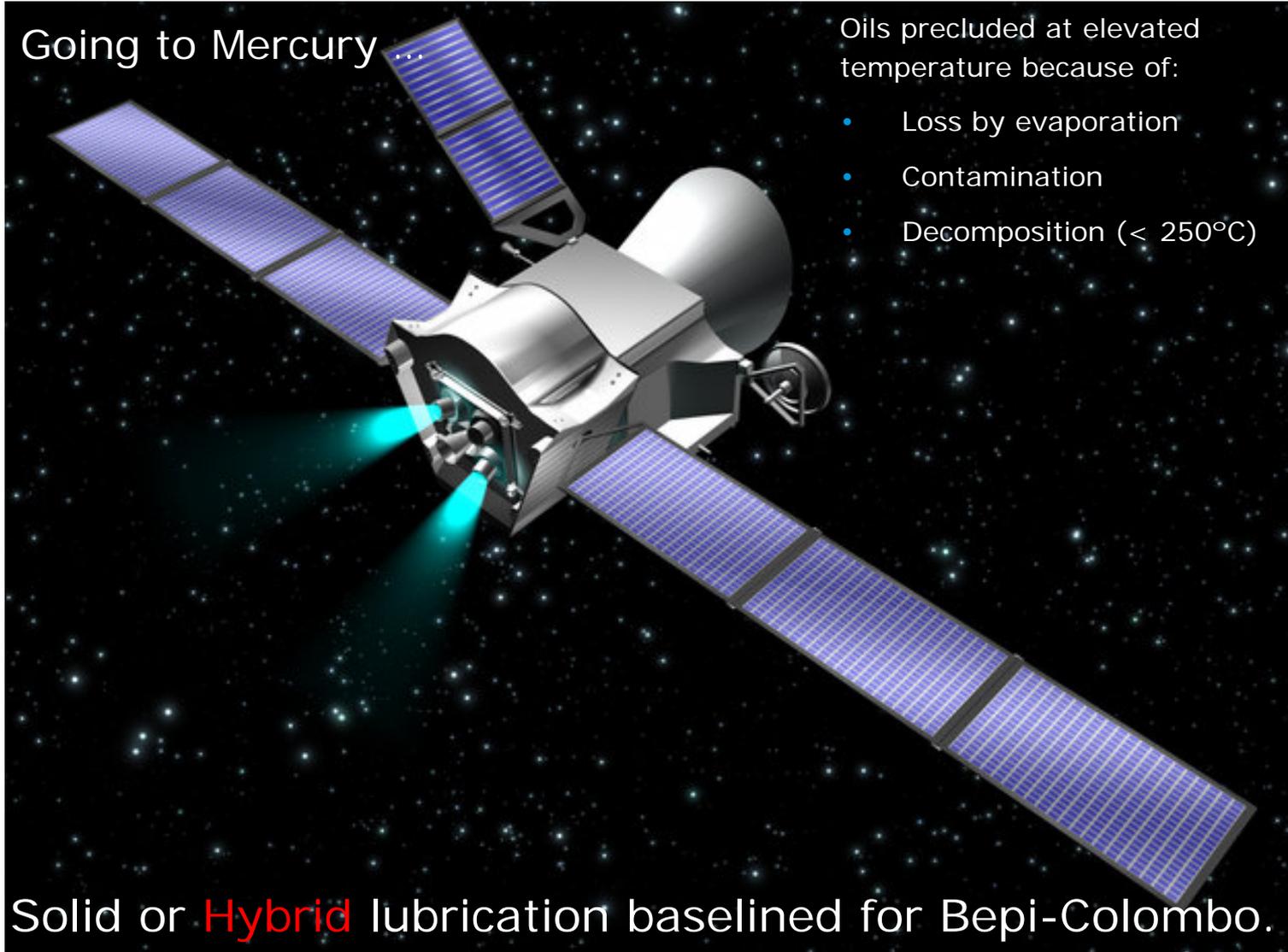
JWST has complex deployment systems and many actuators operating in cryogenic conditions (~30K)

Must use solid lubricants – but unlike Hubble will not be in-flight serviceable.

Solid Lubrication (4K and 20K)



Going to Mercury ...



Oils precluded at elevated temperature because of:

- Loss by evaporation
- Contamination
- Decomposition (< 250°C)

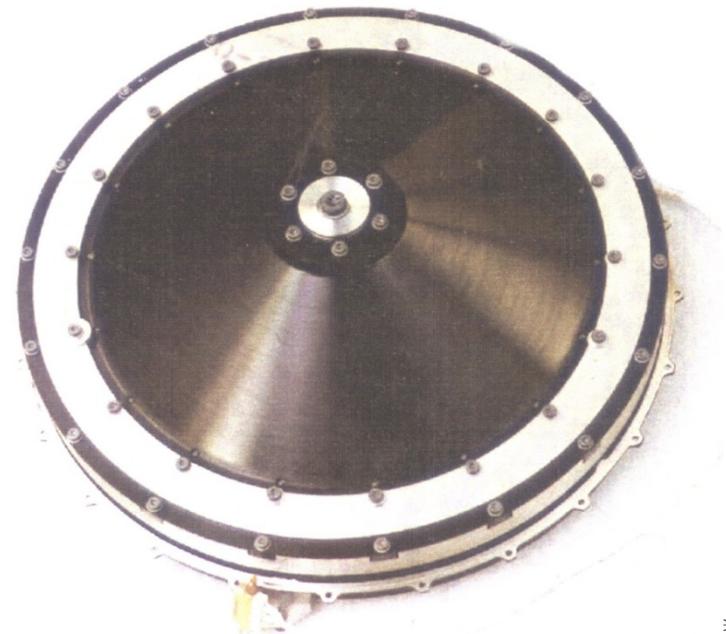
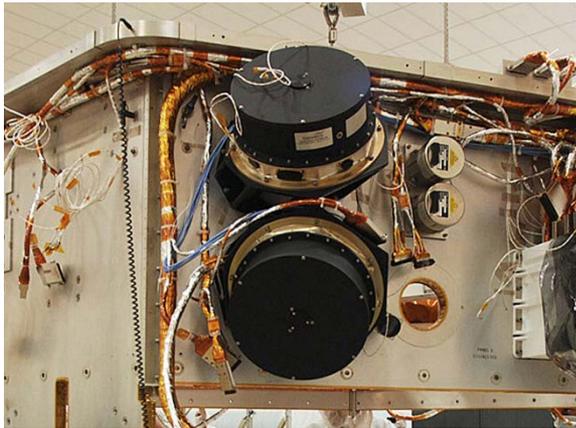
Solid or **Hybrid** lubrication baselined for Bepi-Colombo.

... for highest duty applications

Limit of life for solid lubricated bearings is ~1000 million revs.

Oils in hydrodynamic lubrication mode can last considerably longer e.g. momentum wheels

Important to maintain oil where required... and understand bearing stability

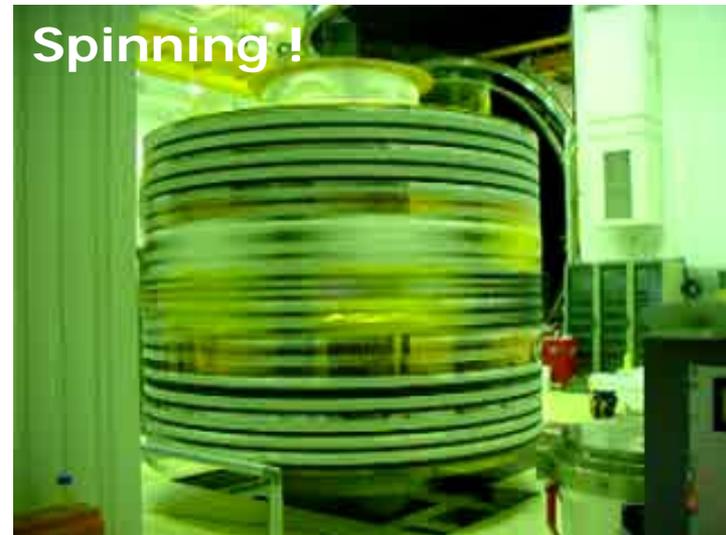
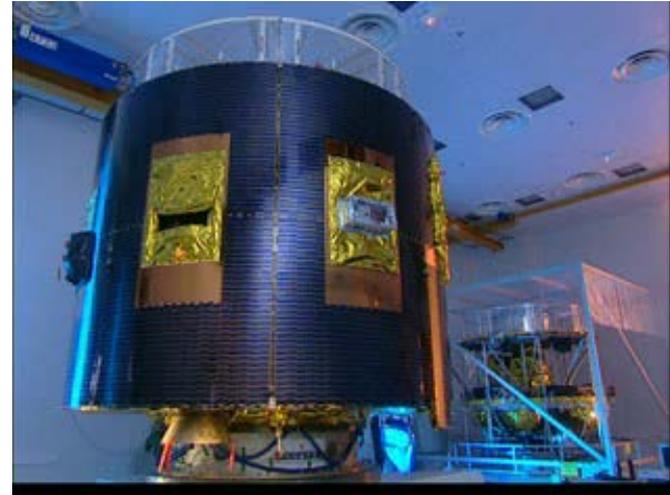


Typical of many spacecraft – lubricants selected to be optimal for the specific requirements of each instrument – meaning:

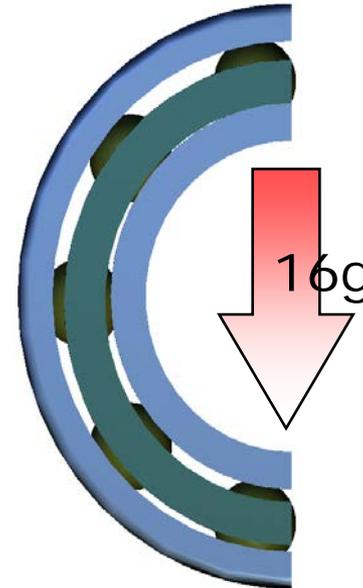
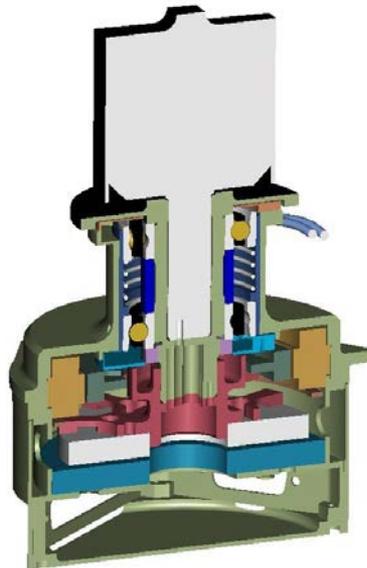
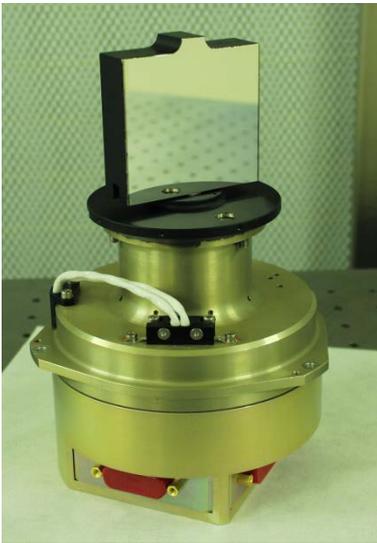
- Instruments:
 - ❖ Lead, MoS₂, self-lubricating bearings, hybrid lubrication (lead and grease), grease alone.
- Reaction Wheels:
 - ❖ Oil

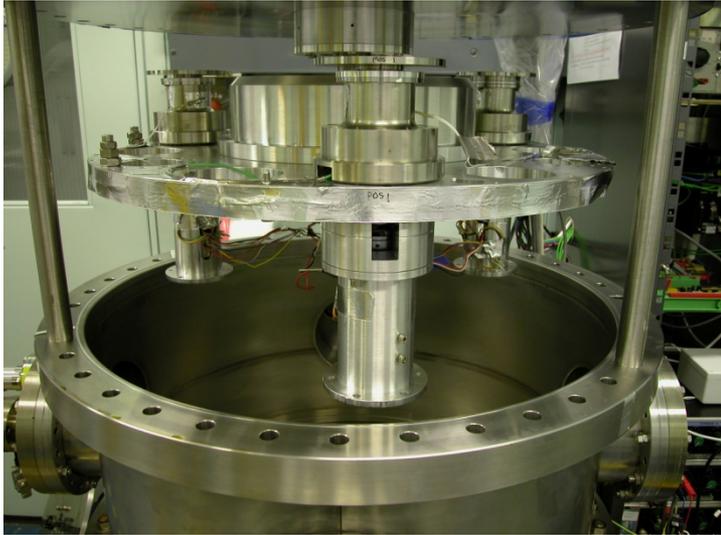


- MSG-1,2 and 3 in GEO Orbit
- Spin stabilised 100rpm
- GERB Instrument –
Geostationary Earth Radiation
Budget
- Solid lubrication strongly
preferred



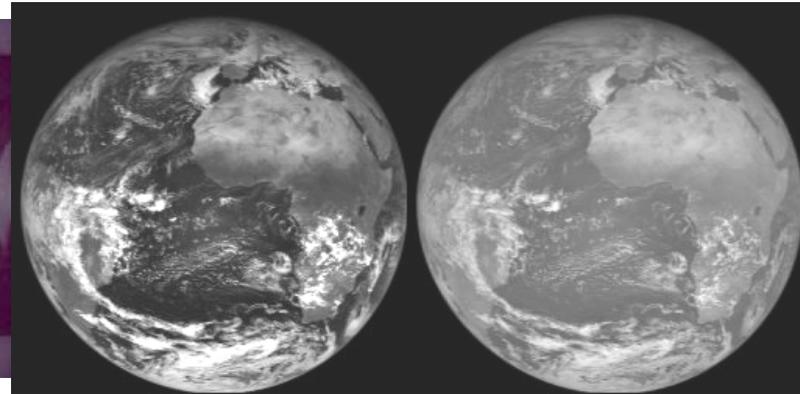
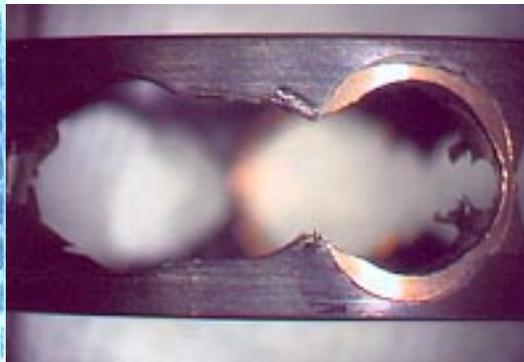
- Spring preloaded bearing pair
- 16g continuous radial acceleration due to position on spacecraft
- Operational requirement 230 million revs



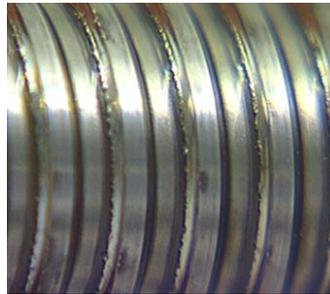
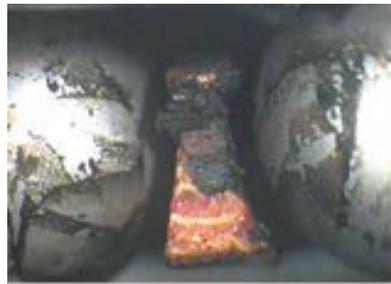
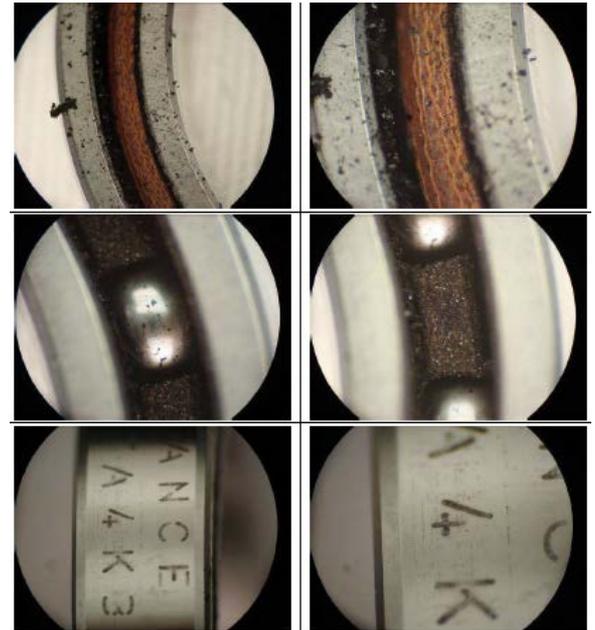
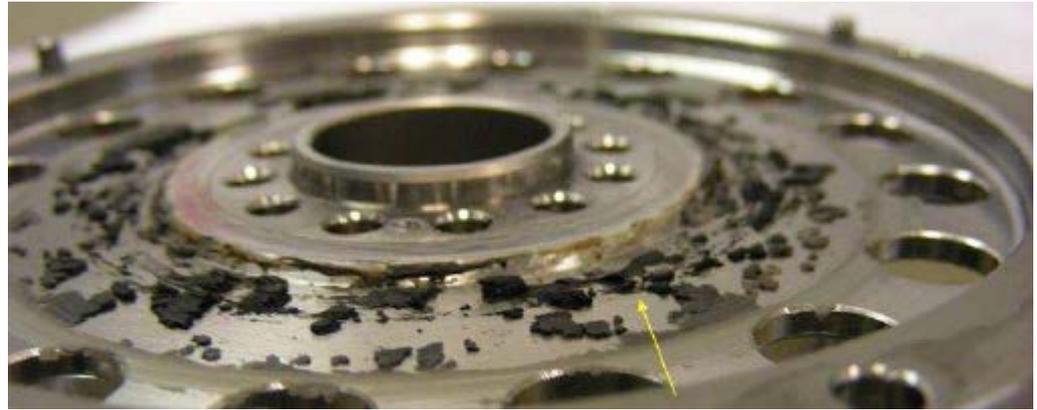


Various concepts trialled
Life-test Under 16g on centrifuge
Design modified to include “Debris traps”

Three major iterations of design
before a successful design was
achieved.



Inspection after Life Test



- Solid coating improvement (MoS₂ process)
 - ❖ Longer life, fewer operational constraints
- New European Self-lubricating cage materials
 - ❖ Lower cost, long life
- New European MAC type of Oil
- Ionic Fluids (a new class of fluid lubricants?)
 - ❖ Lower volatility and greater resistance to degradation than PFPEs
- REACH (Registration, Evaluation and Authorisation of Chemicals)
 - ❖ Replacement of established lubricants (Lead), solvents and other materials

- Long-term storage
 - ❖ Understanding of issues for spacecraft batch assembly, or “in orbit storage” for demisable spacecraft
- European anti-creep barrier
- Re-usable turbomachinery and its implication at tribology levels
- Quest for the Magic Lubricant ???
- No lubricant (magnetic bearings and magnetic gears ...)

- Good application of tribology is critical to successful space missions
- Tribological systems are complex and varied
- It is difficult to predict tribology by “model”; test highly recommended !
- And ... There is still a lot to be done !!!



