

SATELLITE COMMUNICATION WITH OPTICAL WAVES

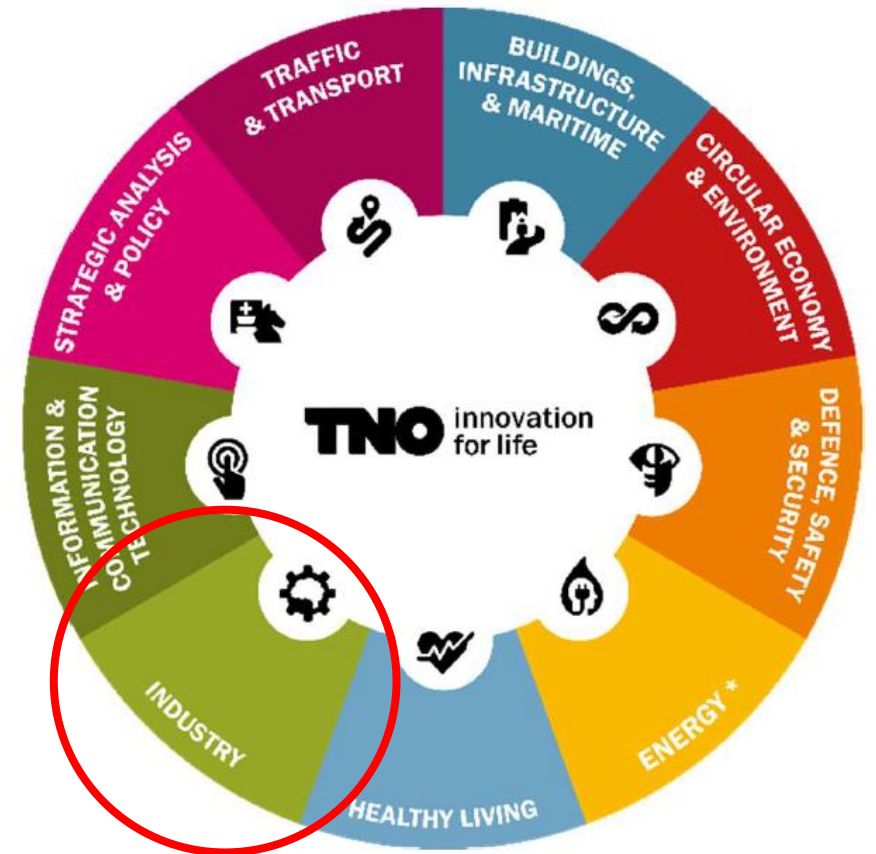
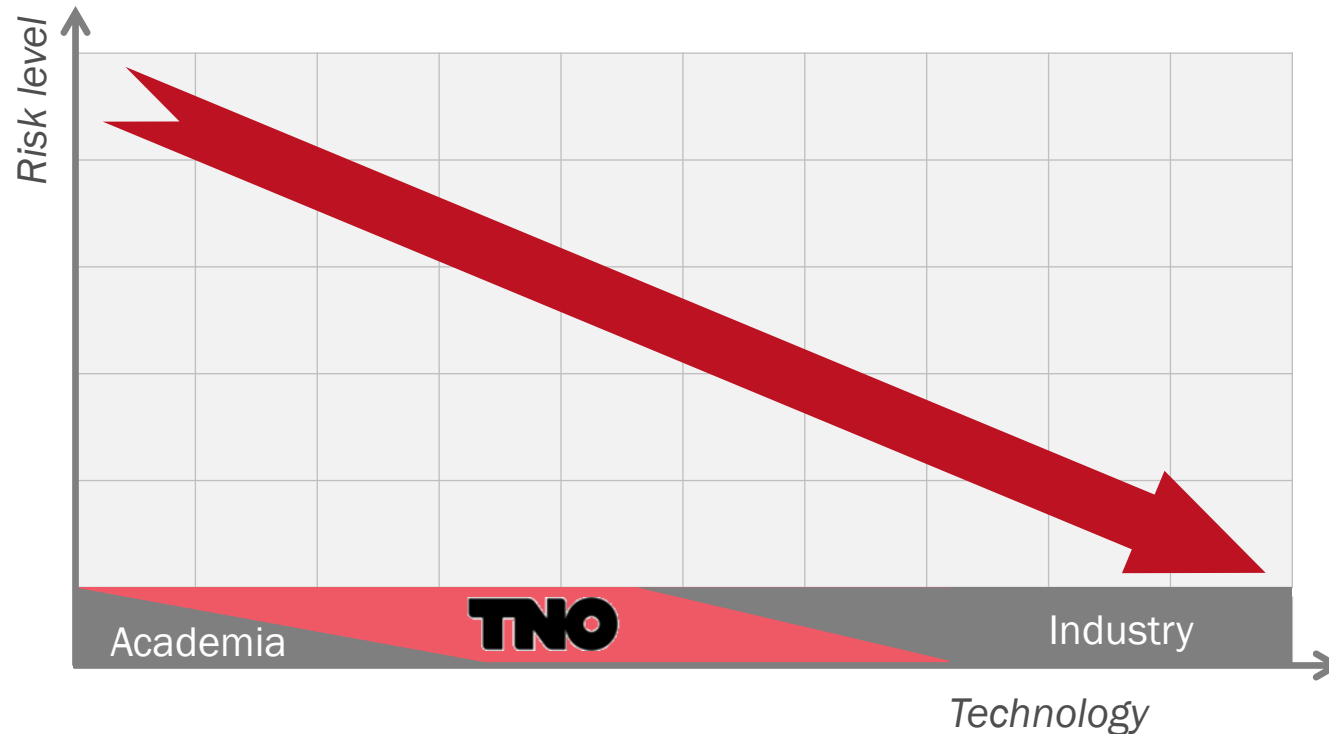
– POTENTIAL AND CHALLENGES –

>
NIEK DOELMAN

- TNO SPACE & LEIDEN UNIVERSITY



- › Founded in 1932
- › Independent not-for-profit organization, bridging gap between academia and industry
- › Annual Turnover ~500Meur: ~30% from Dutch government rest from industry
- › Around 3.000 employees (mainly MSc, PhD)
- › Organized in 9 focus area's/units.



› INDEX

SATELLITE COMMUNICATION WITH OPTICAL WAVES

01. POTENTIAL OF USING OPTICAL WAVES

02. CHALLENGES

03. PRACTICAL EXAMPLES

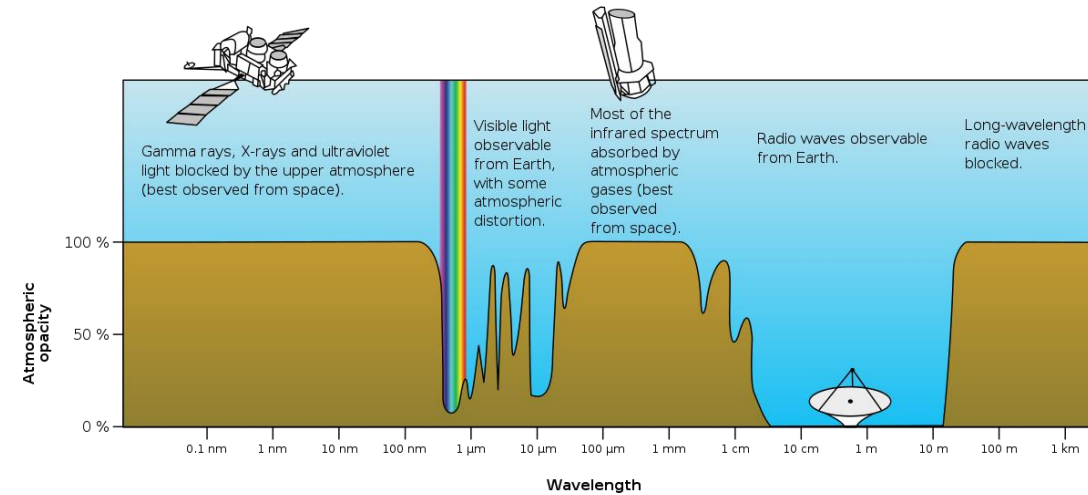
FROM RADIO-FREQUENCY WAVES TO OPTICAL WAVES POTENTIAL

HIGH FREQUENCY RF BANDS

- › EHF 30 - 300 GHz waves: 10 – 1 mm
- › THF 0.3 - 3 THz waves: 1 - 0.1 mm

OPTICAL BANDS

- › 1.064 μm 282 THz
 - › 1.550 μm 194 THz
- } order ~1000 faster/ shorter wavelength



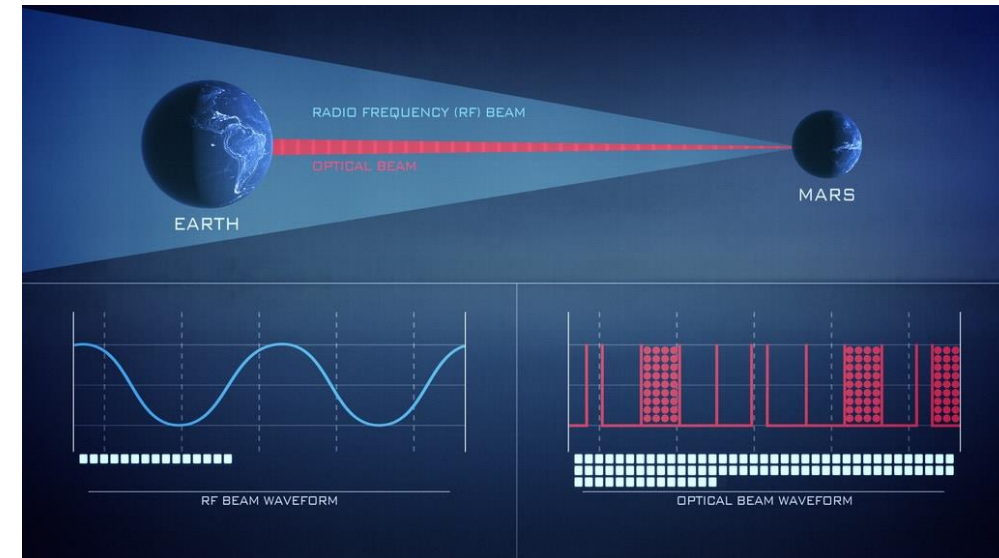
IMPACT

- › Shannon's capacity: Capacity = Bandwidth x $\log_2 \left(1 + \frac{\text{Signal}}{\text{Noise}} \right)$
- › Transmission (Friis): Transmission coefficient $\sim \left(\frac{D_{Tx} D_{Rx}}{\lambda L} \right)^2$

› OPTICAL WIRELESS COMMUNICATION

Main Advantages

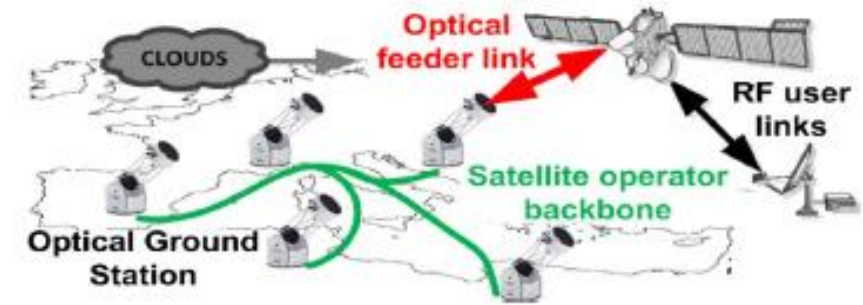
- › High Speed: data rates (> 100 Gbps) with small Transmitter (Tx)/ Receiver (Rx)
- › Secure: Immunity to interception, jamming, interferers and piracy
- › Secure: enables the use of free space Quantum Key Distribution
- › SWaPC: lower volume, lower weight (space), lower cost /bit
- › Bandwidth: abundant, unregulated bandwidth
- › Network: terrestrial fibre optics already available /enabler
- › ... more



Credits: NASA

OPTICAL WIRELESS COMMUNICATION Challenges

- Requires clear Line-of-Sight ... clouds for instance demand a grid of ground stations
- Requires high precision pointing ... much smaller beam spot
- Atmospheric conditions ... absorption, scattering, ... atmospheric turbulence leads to intensity fluctuations, beam wander, beam spread
- Eye safety limits ~50 W/m² (1064 nm) and 1000W/m² (1550 nm)
- Some technology still @ low maturity ... e.g. high-power multiplexing, amplifiers, ..
- Standards missing / in progress
- Additional RF/optics conversion needed
- On-board (satellite) high-speed processing needed
- ...



Poulenard 2015

› FIRST (OPERATIONAL) OPTICAL SATELLITE LINK

› European Data Relay Satellite (EDRS) System

The SpaceDataHighway
The 1st Operational Optical Communication system

- We have achieved more than **50,000** successful relay links since 2016
- Outstanding service availability greater than **99.5 %**
- Up to **39** link sessions per day
- Downloaded more than **3,000,000** GB of data from space
- SpaceDataHighway** can download data from space in near-real-time

What benefits does our services bring to our customers

- Enabling last minute satellite reactivity and tasking
- Real time data access and high volume data transfer
- Improved data latency on a global scale
- Protected comms to support next-gen platforms and multi-domain operations

AIRBUS

- Secure data transfer up to 1.8 Gbps
- Data relay from Low Earth Orbits (700km) to GEO-stationary
- Relay up to 40 TB of data
- Data from Observation satellites, aircraft, unmanned aerial vehicles and other mobile

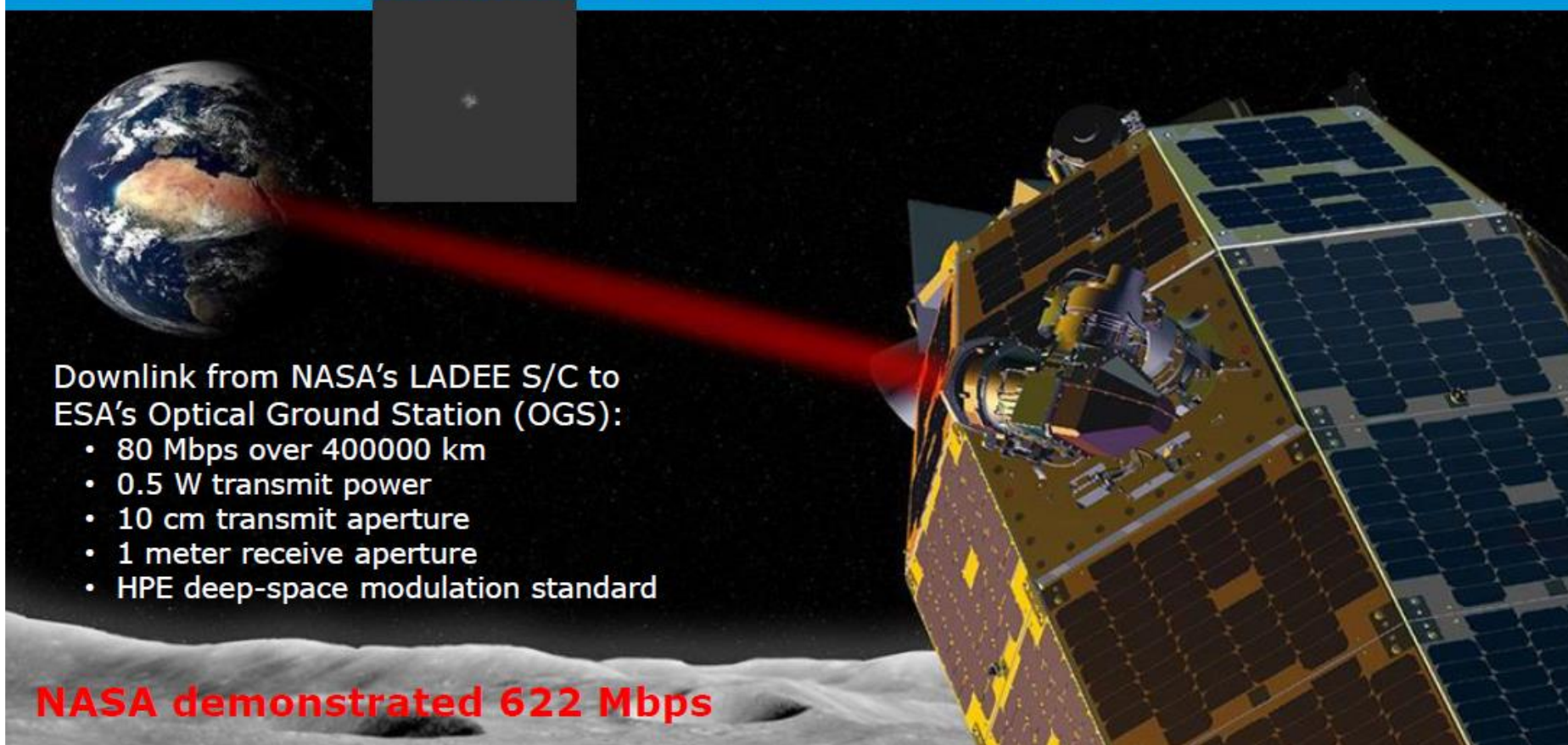
Credits: Airbus 2021

EARLY OPTICAL SATELLITE LINKS

Lunar Link

European Deep-Space Optical Communication Program

Lunar Laser Communication Demonstration (LLCD)



Downlink from NASA's LADEE S/C to
ESA's Optical Ground Station (OGS):

- 80 Mbps over 400000 km
- 0.5 W transmit power
- 10 cm transmit aperture
- 1 meter receive aperture
- HPE deep-space modulation standard

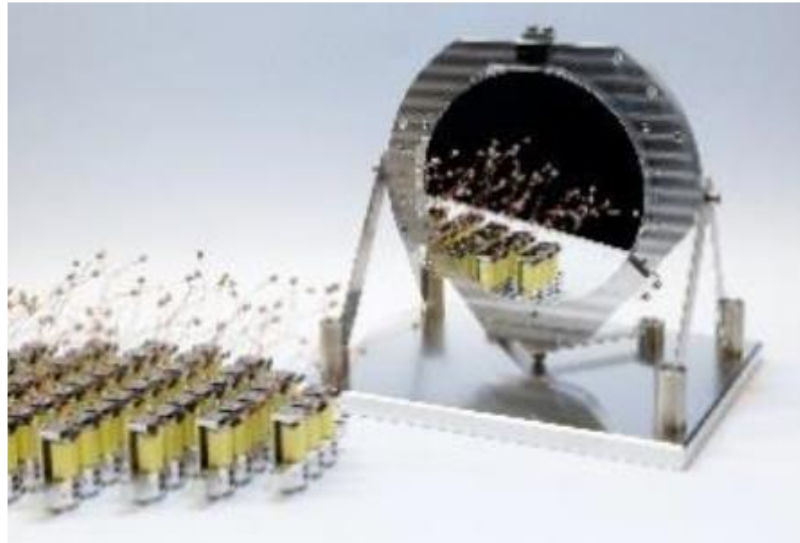
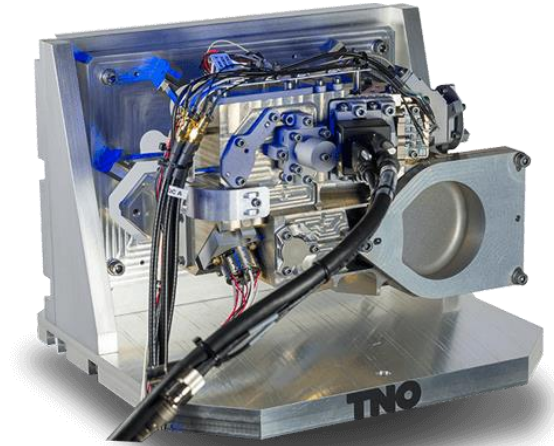
NASA demonstrated 622 Mbps

Credits: ESA

› OPTICAL SATELLITE COMMUNICATION

Details of a Ground to LEO Satellite Link

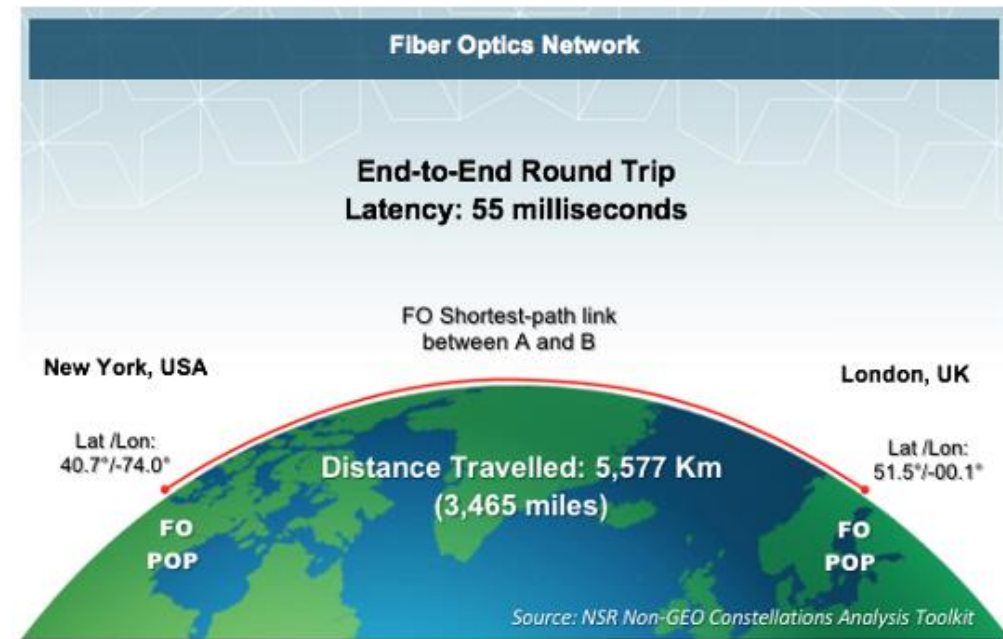
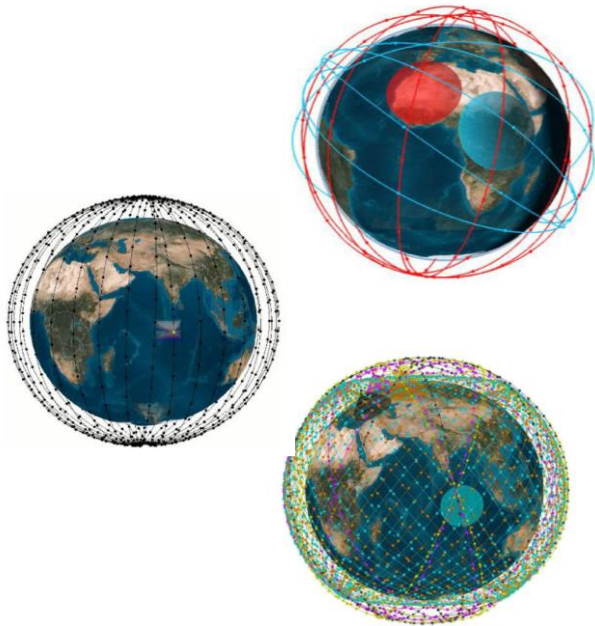
- Specifics of optical wireless over very long distance
- PHY-layer perspective



LEO SATELLITE CONSTELLATIONS

HIGH THROUGHPUT AND LOW LATENCY

- › Various LEO satellite configs under construction ..
- › Objective: high-throughput broadband services with low latency
- › ... latency advantage over fiber network

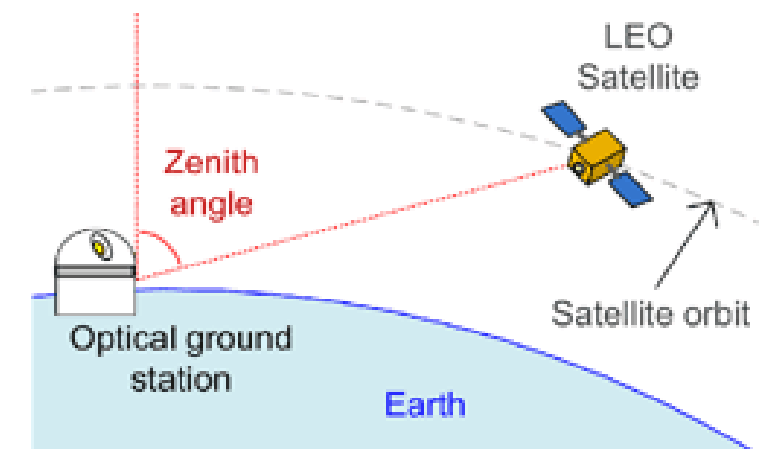


› SATELLITE - GROUND LINK

DOWNLINK BUDGET

LINK to LEO Satellite

Geometrical	Tx Antenna Gain
	Rx Receiver Gain
	Free-space Loss
	TOTAL Geometric
Optical inaccuracies	Optical aberrations in Tx
	Optical aberrations in Rx
	Pointing Boresight error
	Pointing Precision
	TOTAL Inaccuracies
Atmospheric	Absorption/ scattering
	Wavefront error
	Irradiance fluctuations
	TOTAL Atmospheric
Total	



Ref: Takemoto 2015 ICSOS

› Free-space Loss

› Range effect: $L_R = \left(\frac{\lambda}{4\pi L}\right)^2$... -252 dB (Zenith) -260 dB (20deg elevation)

› LEO satellite at 500 km orbit, 1550 nm wave

› Antenna Gains

› Gain: $G \approx \left(\frac{\pi D}{\lambda}\right)^2$ ground: +126 dB space: +110 dB

› Limited apertures: ground: 0.5 – 1 m (very large: 4m),

space: 0.1 - 0.2 m (very large: 1m)

› SATELLITE - GROUND LINK

DOWNLINK BUDGET

LINK to LEO Satellite

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Total	

› Optical wavefront errors

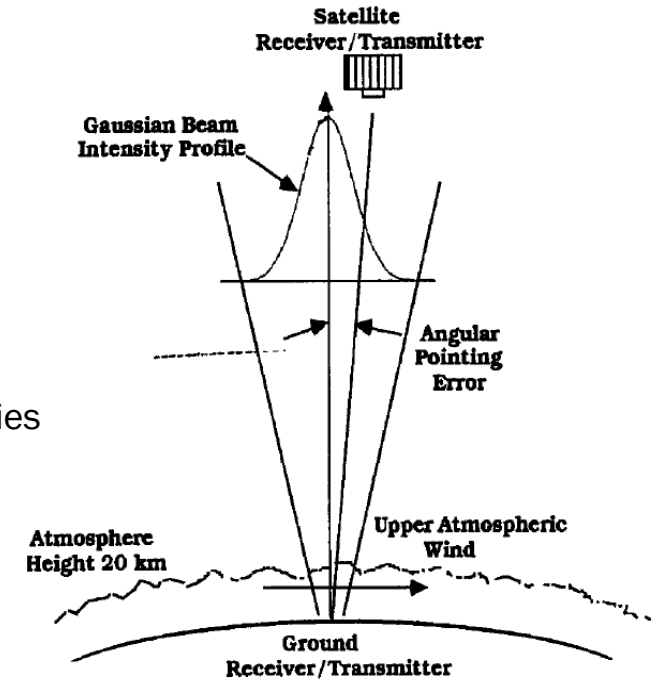
- › Distorted beams, due to optical component inaccuracies
- › Wavefront error of 200nm, leads to ~3 dB loss

› Pointing issues

- › Boresight error: ..slope of the Gaussian...
- › and jitter: ..spot ‘dancing’ in receiver plane

› Absorption and scattering effects / molecular and aerosol

- › Extinction factor
- › Scales with ‘number density of air constituents’
- › Integrated effect (over path)
- › Typical value: 1-2 dB



Ref: Andrews 1995

Ref: Kaushal 2017

$$\gamma(\lambda) = \underbrace{\alpha_m(\lambda)}_{\text{Molecular absorb. coeff.}} + \underbrace{\alpha_a(\lambda)}_{\text{Aerosol absorb. coeff.}} + \underbrace{\beta_m(\lambda)}_{\text{Molecular scatt. coeff.}} + \underbrace{\beta_a(\lambda)}_{\text{Aerosol scatt. coeff.}}$$

IMPACT OF ATMOSPHERIC TURBULENCE

BEAM DISTORTION

- Temperature/ pressure variations lead to fluctuations (spatial & temporal) in index-of-refraction
 - Cn2 profile – strongest at ground level
 - Power spectrum – fast fall-off for higher spatial frequencies
- Giving pathlength fluctuations → wavefield phase fluctuations
- and amplitude fluctuations (after some propagation)

EFFECTS ON OPTICAL BEAM

- Beam spread; wider spot @ receiver
 - On top of diffractive effect
- On-axis scintillation (amplitude fluctuations)
- Beam wander; spot ‘dancing’ in receiver plane
 - also yielding intensity fluctuations

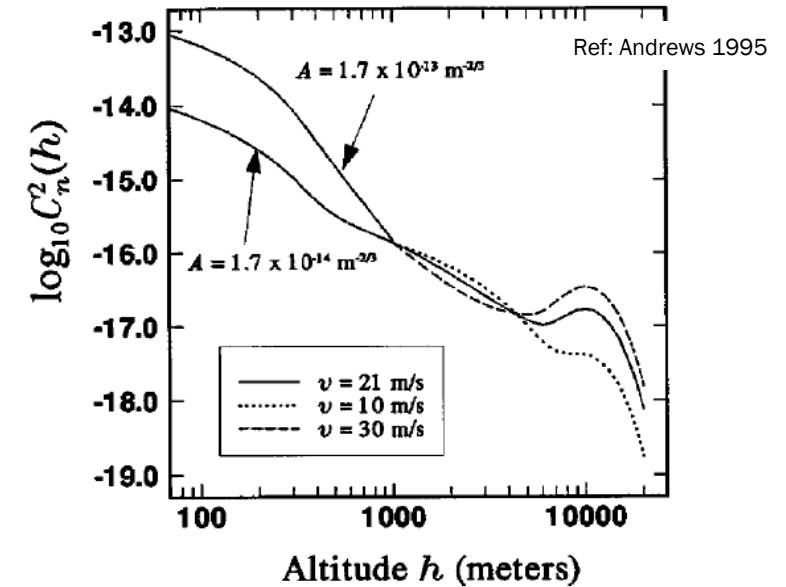
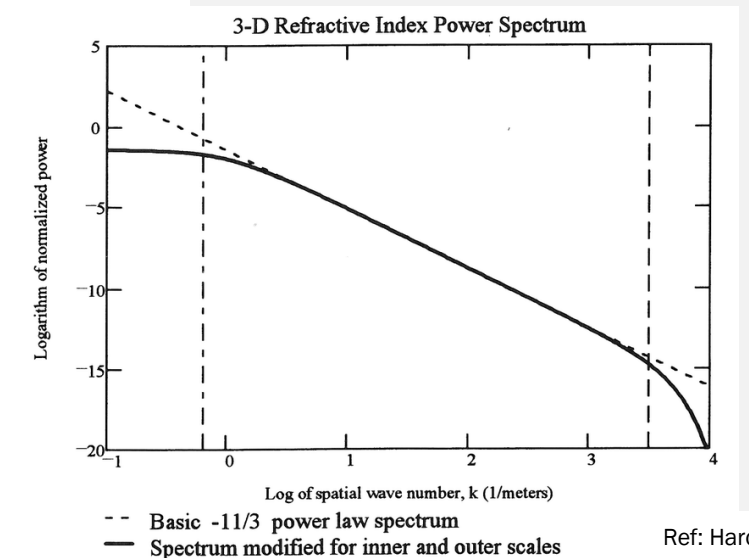


Fig. 2. $C_n^2(h)$ profile associated with the H-V day model as a function of altitude h . The values of A shown represent C_n^2 near ground level, whereas v denotes high-altitude wind speed.



ATMOSPHERIC TURBULENCE

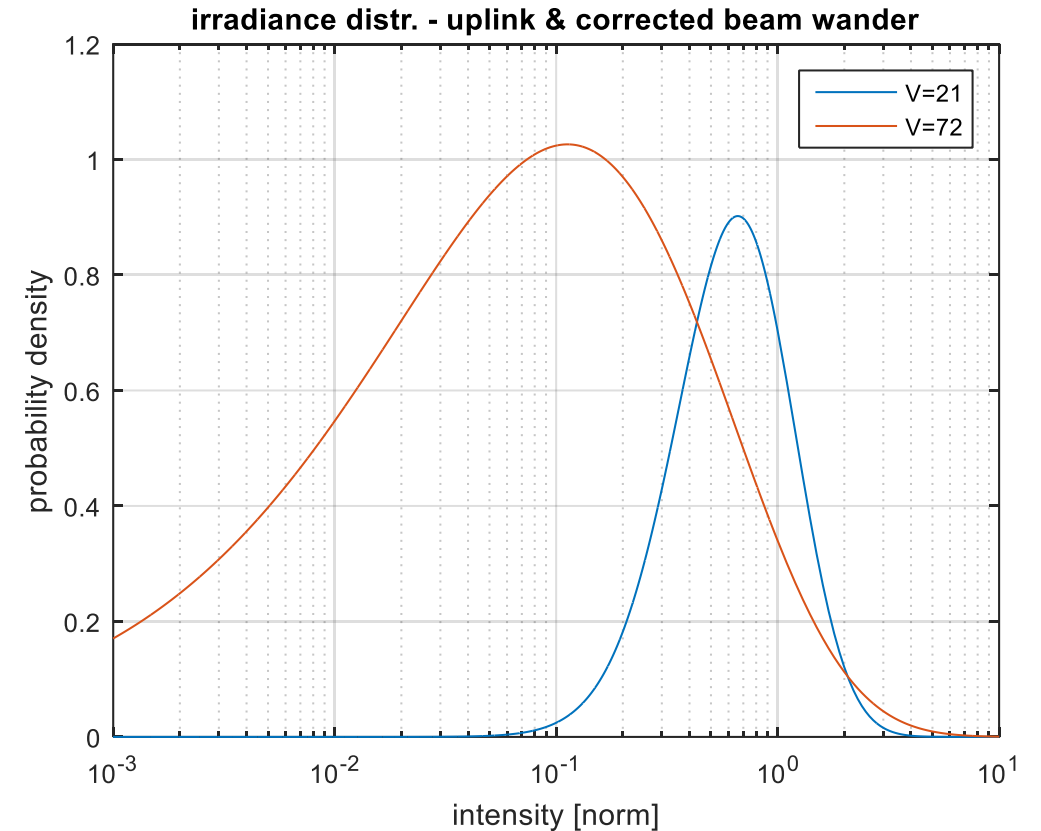
EFFECT ON LINK BUDGET

1. Average Loss
2. Instantaneous intensity fluctuations
(fading channel)

IMPORTANT PARAMETERS

1. Turbulence strength
2. Elevation angle
3. Slant range
4. Uplink vs. downlink
5. Beam size/ type

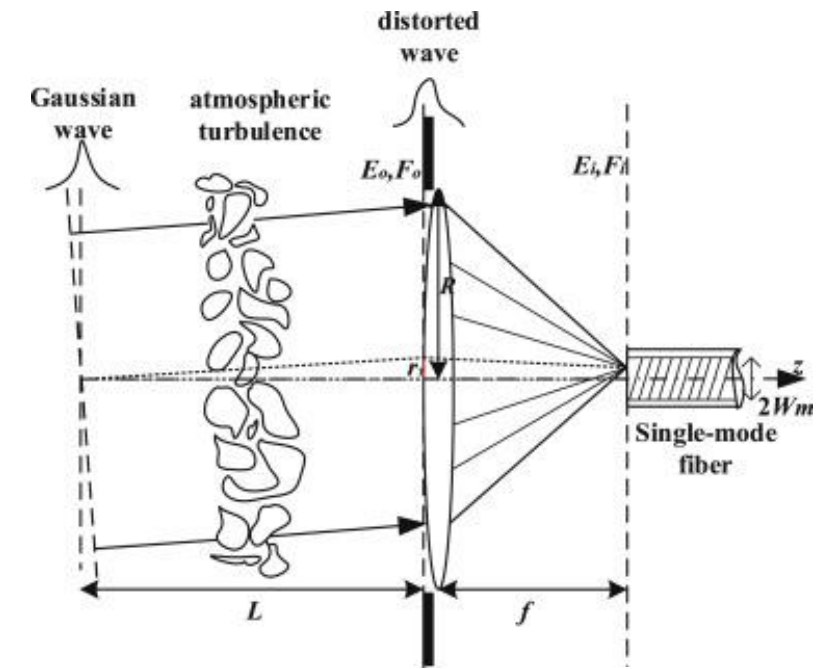
Model of intensity fluctuations



- Cn2 Model: Hufnagel-Valley with variable pseudo-wind at 11 km
- Values: [21 72]
- Uplink to satellite
- Other: Tx waist (0.35m), Zenith angle (1 rad), wavelength (1550 nm), AO beam wander correction under PAA angle

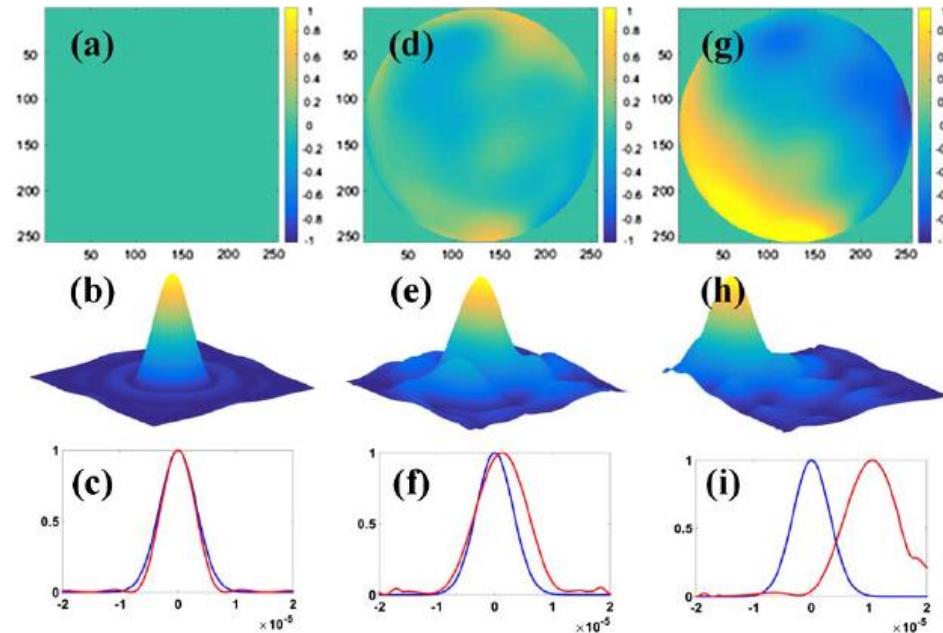
› DISTURBED WAVEFRONTS AND FIBER COUPLING

1. Wave intensity profile should match mode of fiber
2. Wave distortion may seriously reduce coupling efficiency
3. Nominal case: coupling **efficiency of 0.81**
 - Spatial Match of Gaussian wave profile with fiber mode



Ref: Lu 2019

WAVEFRONT PHASE CORRECTION NECESSARY !



Ref: Chen 2015

Fig. 4. Schematic of the fiber coupling over different turbulence.

ADAPTIVE OPTICS

WORKING PRINCIPLES

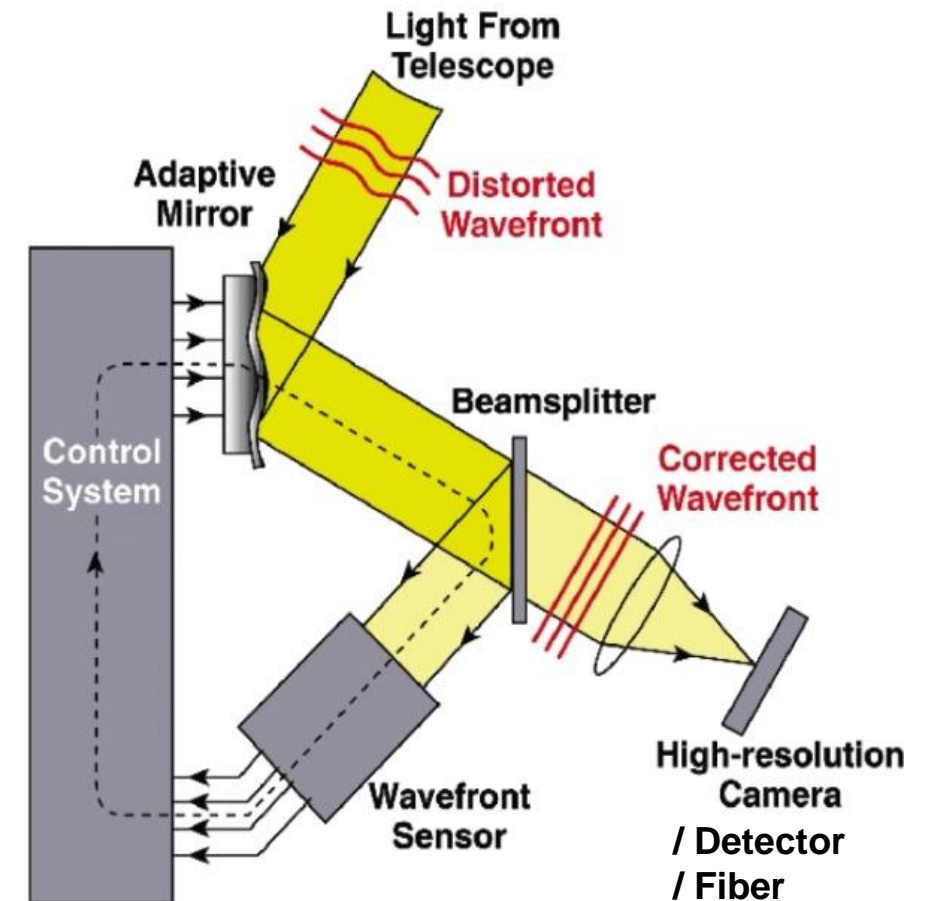
PRINCIPLES

1. Measure phase distortion at receiver
2. Compute (instantaneous) wavefront correction
3. Apply phase correction by 2D deformable mirror

PROPERTIES

1. Real-time feedback loop
2. 2D spatial correction for a 3D effect
3. Relatively complex solution, with high-end components

(indispensable in large astronomical telescopes)

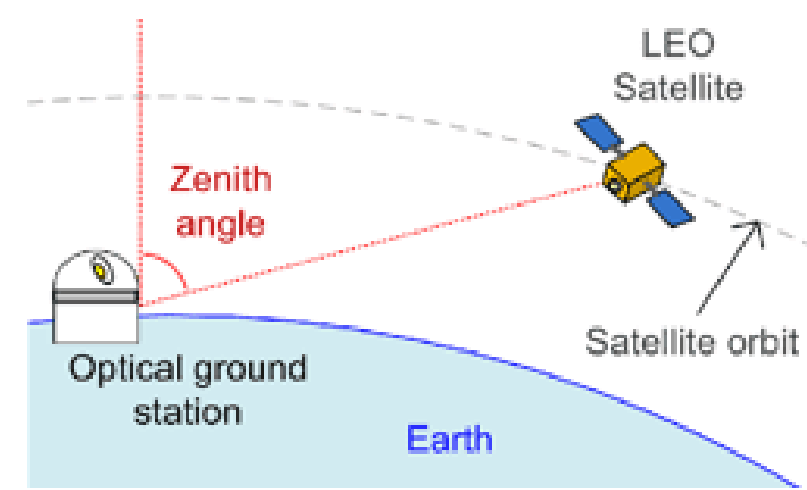


Movie: https://www.youtube.com/watch?v=3BpT_tXYy_I
from 1m20

› SATELLITE TO GROUND LINK DOWNLINK BUDGET

LINK Down LEO Satellite (500 km) – 1550 nm - **Zenith** – Rx 0.4m

Geometrical	Tx Antenna Gain	109.7	Tx aperture = 0.15m / Beam divergence 9.2 μ rad (half angle)
	Rx Receiver Gain	118.2	Rx aperture = 0.4 m
	Free-space Loss	-252.2	Slant range = 500 km
	TOTAL Geometric	-24.3	
Optical inaccuracies	Optical aberrations in Tx	-0.7	Wavefront Error = 100 nm (RMS)
	Optical aberrations in Rx	-1.2	Transmission Rx optics system = 0.75
	Pointing Boresight error	-0.6	Offset = 2.5 μ rad
	Pointing Precision	-1.1	Jitter is 2.5 μ rad (RMS)
	TOTAL Inaccuracies	-3.7	
Atmospheric	Absorption/ scattering	-0.9	Visibility = 20 km, Scale parameter = 4km
	Wavefront error	-1.8	Mismatch for fiber coupling & nominal factor 0.81 / with Adaptive Optics system at 75% rejection
	Irradiance fluctuations	-0.5	Margin for outage probability of 1%
	TOTAL Atmospheric	-3.2	
Total		-31.2 dB	

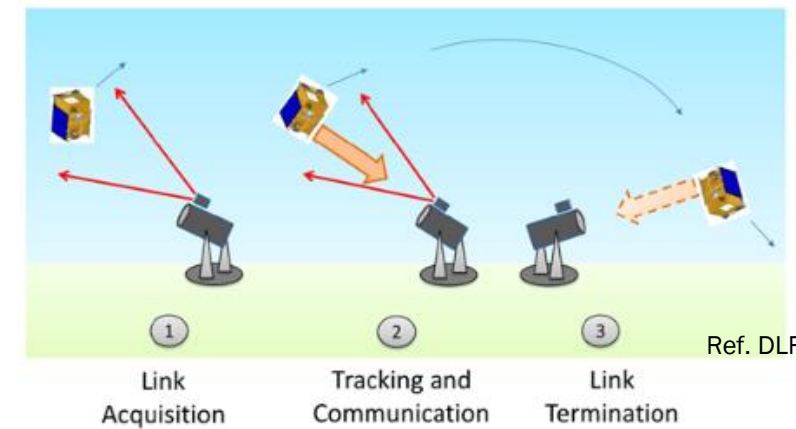


SATELLITE TO GROUND LINK

DOWNLINK BUDGET

LINK Down LEO Satellite (500 km) – 1550 nm - extreme angles – Rx 0.4m

		Zenith	20 deg elevation	
Geometrical	Tx Antenna Gain	109.7	109.7	Tx aperture = 0.15m / Beam divergence 9.2 μ rad (half angle)
	Rx Receiver Gain	118.2	118.2	Rx aperture = 0.4 m
	Free-space Loss	-252.2	-259.7	Slant range = 500 km vs. 1192 km (Orbit at 500km)
	TOTAL Geometric	-24.3	-31.9	7.6 dB higher loss
Optical inaccuracies	Optical aberrations in Tx	-0.7	-0.7	Wavefront Error = 100 nm (RMS)
	Optical aberrations in Rx	-1.2	-1.2	Transmission Rx optics system = 0.75
	Pointing Boresight error	-0.6	-0.6	Offset = 2.5 μ rad
	Pointing Precision	-1.1	-1.1	Jitter is 2.5 μ rad (RMS)
	TOTAL Inaccuracies	-3.7	-3.7	unchanged
Atmospheric	Absorption/ scattering	-0.9	-2.6	Visibility = 20 km, Scale parameter = 4km
	Wavefront error	-1.8	-3.6	Longer atmospheric path/ Adaptive Optics system at 75%
	Irradiance fluctuations	-0.5	-2.4	Margin for outage probability of 1%
	TOTAL Atmospheric	-3.2	-8.6	5.4 dB higher loss
Total		-31.2 dB	-44.2 dB	13 dB higher loss

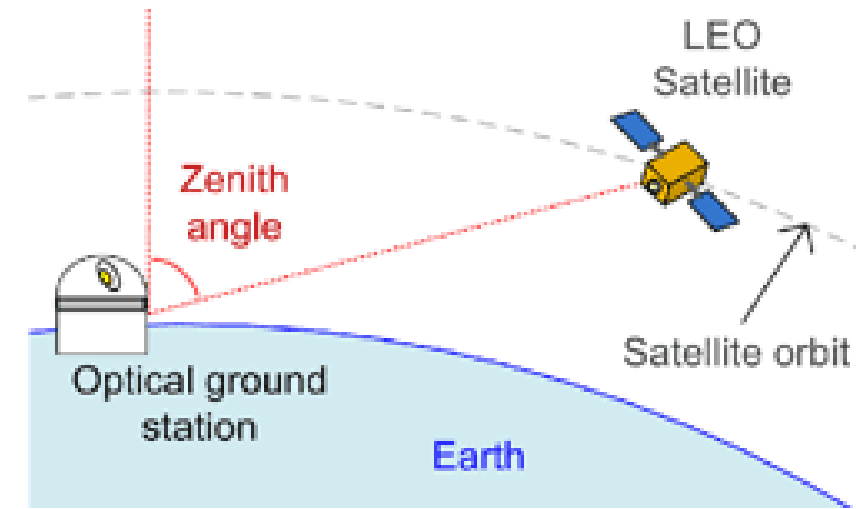


Main effects:

- Longer path
- More atmospheric distortion

› SATELLITE TO GROUND LINK

DOWNLINK BUDGET – INCREASED RX APERTURE



	Rx 0.4m		Rx 0.8 m		
	Zenith	20 deg elevation	Zenith	20 deg elevation	
Tx Antenna Gain	109.7	109.7	109.7	109.7	Tx aperture = 0.15m / Beam divergence 9.2 μrad (half angle)
Rx Receiver Gain	118.2	118.2	124.2	124.2	Rx aperture = 0.8 m
Free-space Loss	-252.2	-259.7	-252.2	-259.7	Slant range = 500 km vs. 1192 km (Orbit at 500km)
TOTAL Geometric	-24.3	-31.9	-18.3	-25.9	<i>6 dB gain with twice the diameter</i>
Optical aberrations in Tx	-0.7	-0.7	-0.7	-0.7	Wavefront Error = 100 nm (RMS)
Optical aberrations in Rx	-1.2	-1.2	-1.2	-1.2	Transmission Rx optics system = 0.75
Pointing Boresight error	-0.6	-0.6	-0.6	-0.6	Offset = 2.5 μrad
Pointing Precision	-1.1	-1.1	-1.1	-1.1	Jitter is 2.5 μrad (RMS)
TOTAL Inaccuracies	-3.7	-3.7	-3.7	-3.7	<i>unchanged</i>
Absorption/ scattering	-0.9	-2.6	-0.9	-2.6	Visibility = 20 km, Scale parameter = 4km
Wavefront error	-1.8	-3.6	-3.9	-9.4	Higher error variance ($D^{5/3}$) / (<i>Adaptive Optics at 75%</i>)
Irradiance fluctuations	-0.5	-2.4	-0.2	-1.1	Effect of Aperture Averaging
TOTAL Atmospheric	-3.2	-8.6	-5.0	-13.1	<i>up to 4.5 dB higher loss</i>
TOTAL	-31.2 dB	-44.2 dB	-27.0 dB	-42.7 dB	<i>Limited gain for low elevation angles</i>

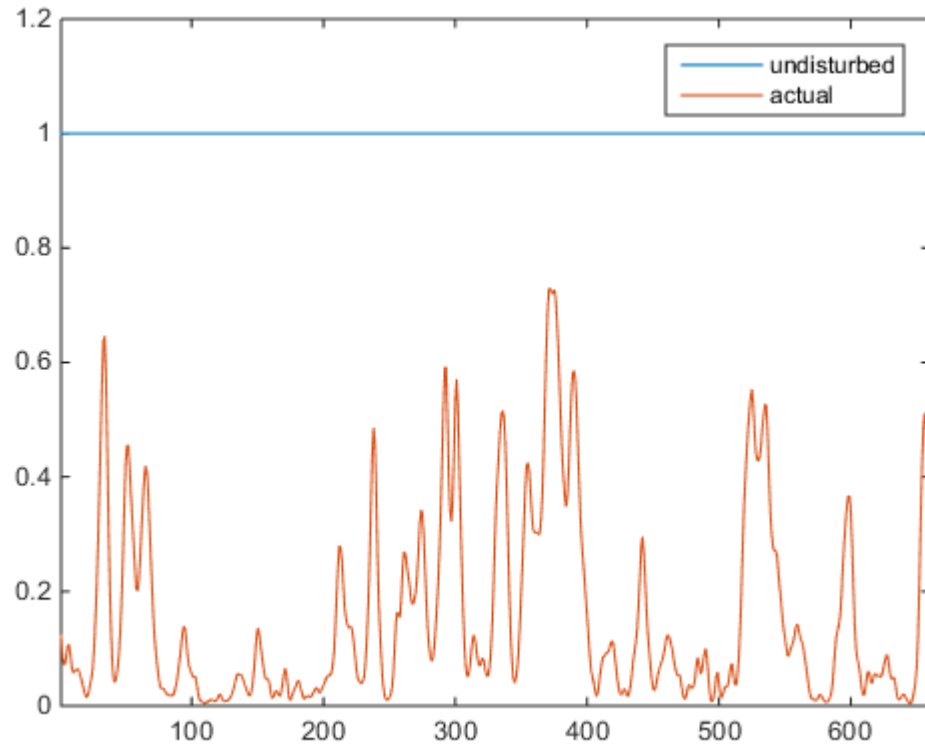
Main effects:

- Receiver gain
- More atmospheric distortion at low elevations

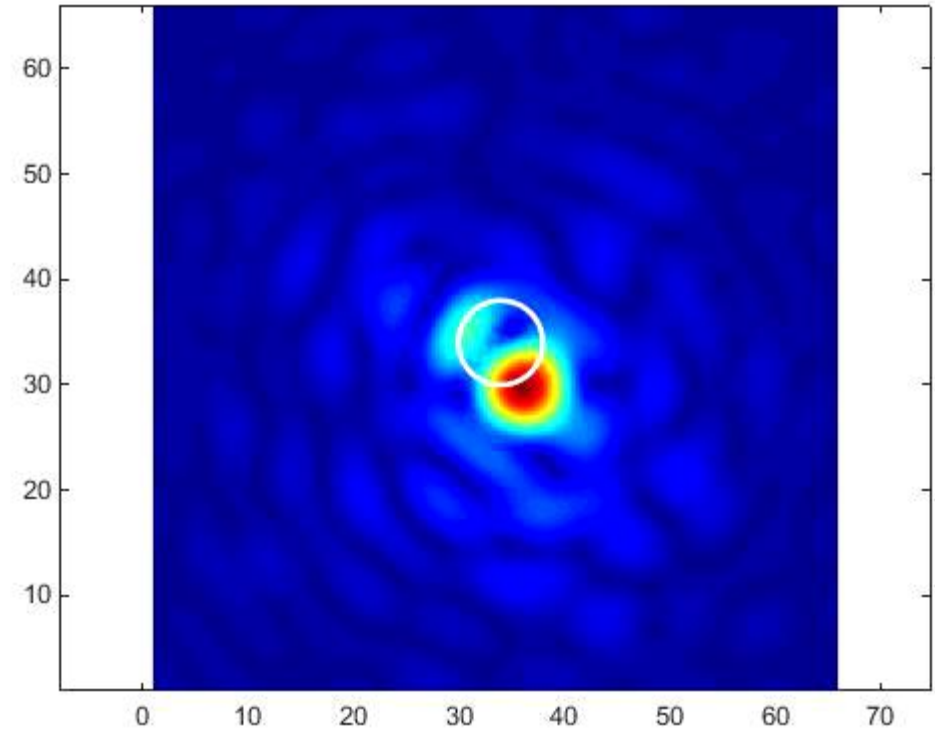
TURBULENCE INDUCED DISTORTIONS

ROLE OF RECEIVER APERTURE

1. Small Rx aperture



Received power

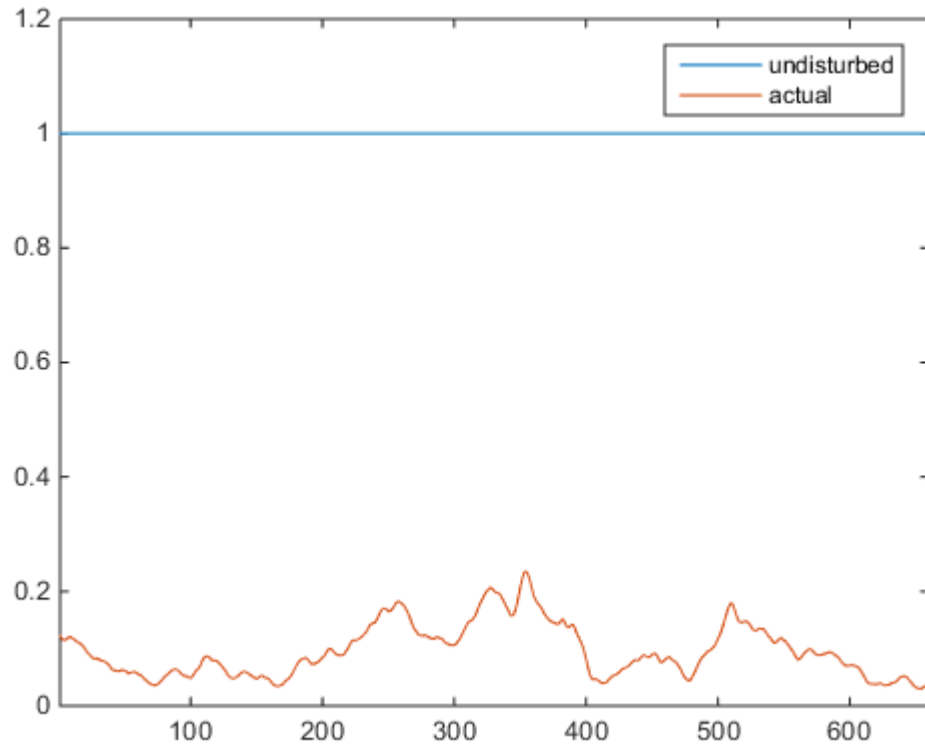


Disturbed spot (atmospheric turbulence)

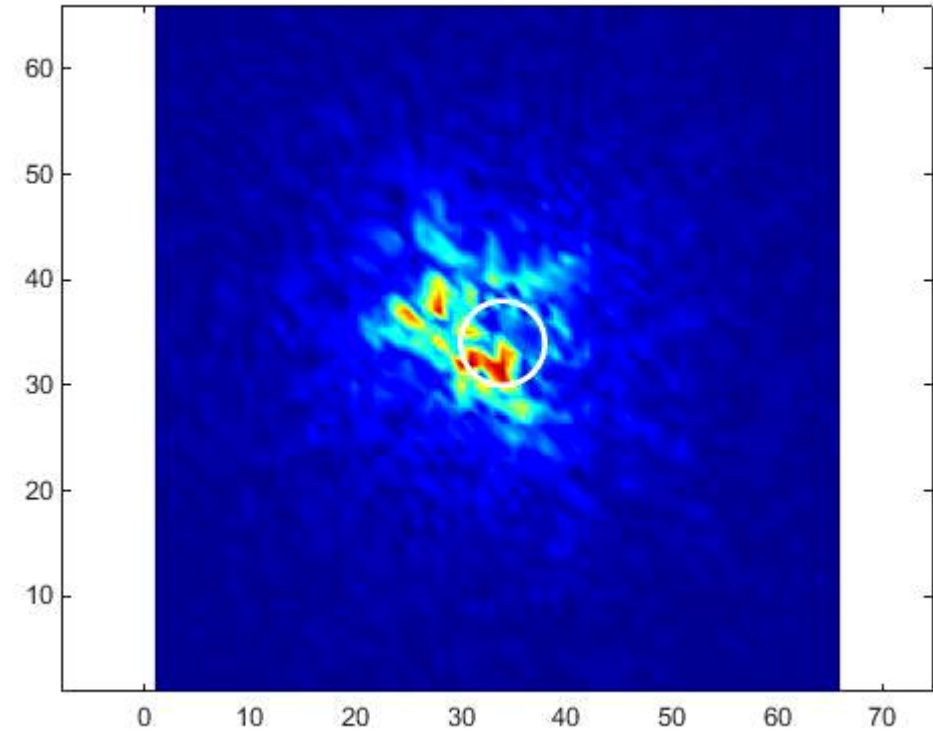
› TURBULENCE INDUCED DISTORTIONS

ROLE OF RECEIVER APERTURE

2. 4 times larger Rx aperture



Received power



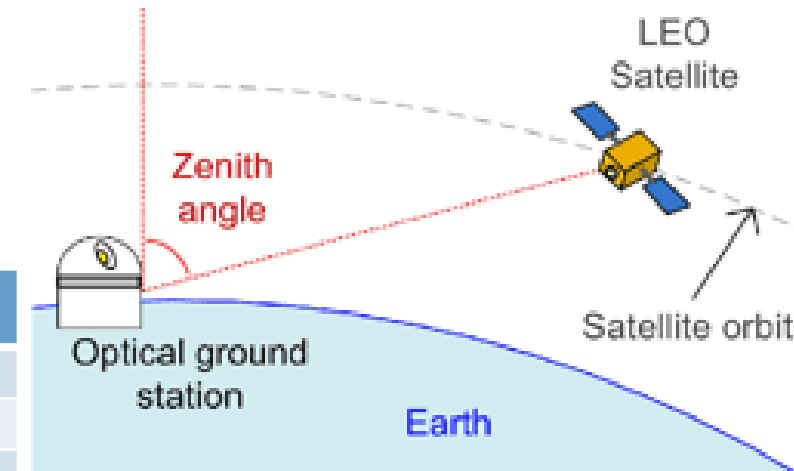
Disturbed spot
(atmospheric turbulence)

GROUND TO SATELLITE LINK

UPLINK BUDGET – regular sized Tx (0.8m) and Rx (0.15m)

LINK to LEO Satellite (500 km) – 1550 nm - Zenith angle

Geometrical	Tx Antenna Gain	124.2	Tx aperture = 0.8 m
	Rx Receiver Gain	109.7	Rx aperture = 0.15 m
	Free-space Loss	-252.2	Slant range = 500 km
	TOTAL Geometric	-18.3	<i>unchanged</i>
Optical inaccuracies	Optical aberrations in Tx	-0.7	Wavefront Error = 100 nm (RMS)
	Optical aberrations in Rx	-1.2	Transmission Rx optics system = 0.75
	Pointing Boresight error	-17.8	Offset = 2.5 μ rad- Tx gain dependent
	Pointing Precision	-9.6	Jitter is 2.5 μ rad (RMS) – Tx gain dependent
	TOTAL Inaccuracies	-29.4	<i>Pointing sensitivity</i>
Atmospheric	Absorption/ scattering	-0.9	Visibility = 20 km, Scale parameter = 4km
	Wavefront error	-	<i>Irrelevant, due to high coherence radius (>> 1m)</i>
	Additional Beam Spread	-15.6	Turbulence-induced widening of the beam
	Irradiance fluctuations	-10.8	100% Pre-corrected Beam wander
	TOTAL Atmospheric	-27.2	
Total		-75.0 dB	<i>~ 48 dB additional loss due to Atmospheric distortion and Pointing errors</i>



Main effects of reversing Tx Rx size

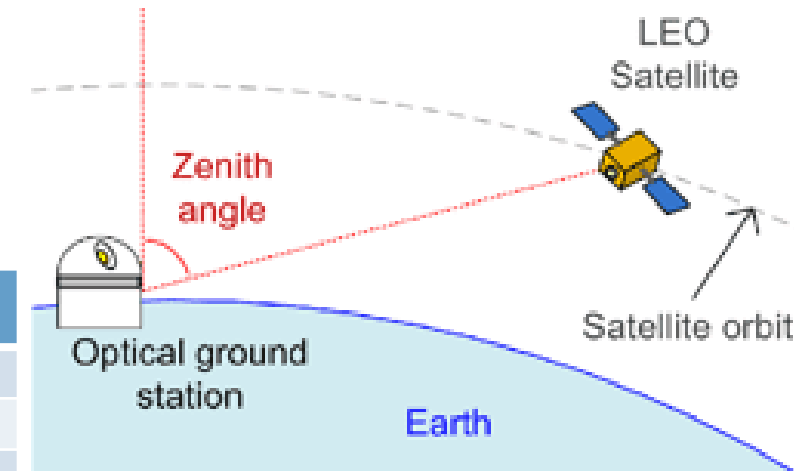
- Same geometric loss
- Very sensitive to pointing errors
- High impact of turbulence

GROUND TO SATELLITE LINK

UPLINK BUDGET – reduced Tx (0.2m) and Rx (0.15m)

LINK to LEO Satellite (500 km) – 1550 nm - Zenith angle

Geometrical	Tx Antenna Gain	112.2	Tx aperture = 0.2 m
	Rx Receiver Gain	109.7	Rx aperture = 0.15 m
	Free-space Loss	-252.2	Slant range = 500 km
	TOTAL Geometric	-30.3	12 dB higher loss
Optical inaccuracies	Optical aberrations in Tx	-0.7	Wavefront Error = 100 nm (RMS)
	Optical aberrations in Rx	-1.2	Transmission Rx optics system = 0.75
	Pointing Boresight error	-1.1	Offset = 2.5 μ rad
	Pointing Precision	-1.8	Jitter is 2.5 μ rad (RMS)
	TOTAL Inaccuracies	-4.9	
Atmospheric	Absorption/ scattering	-0.9	Visibility = 20 km, Scale parameter = 4km
	Wavefront error	-	Irrelevant, due to high coherence radius (>> 1m)
	Additional Beam Spread	-5.5	Turbulence-induced widening of the beam
	Irradiance fluctuations	-0.8	100% Pre-corrected Beam wander
	TOTAL Atmospheric	-7.2	
Total		-42.4 dB	~33 dB better with a 4 x smaller Tx



Main effects smaller Tx

- Lower gain
- Less sensitive to pointing errors
- Smaller impact of turbulence

REDUCING THE EFFECT OF TURBULENCE

INTENSITY FLUCTUATIONS / FADING

POTENTIAL APPROACHES

1. Aperture Averaging

- Increase of (ground-) Rx aperture reduces intensity fluctuations, b

2. Diversity / spatial and or temporal

- For instance: multiple Tx beams to average out fading effects

3. Adaptive Optics

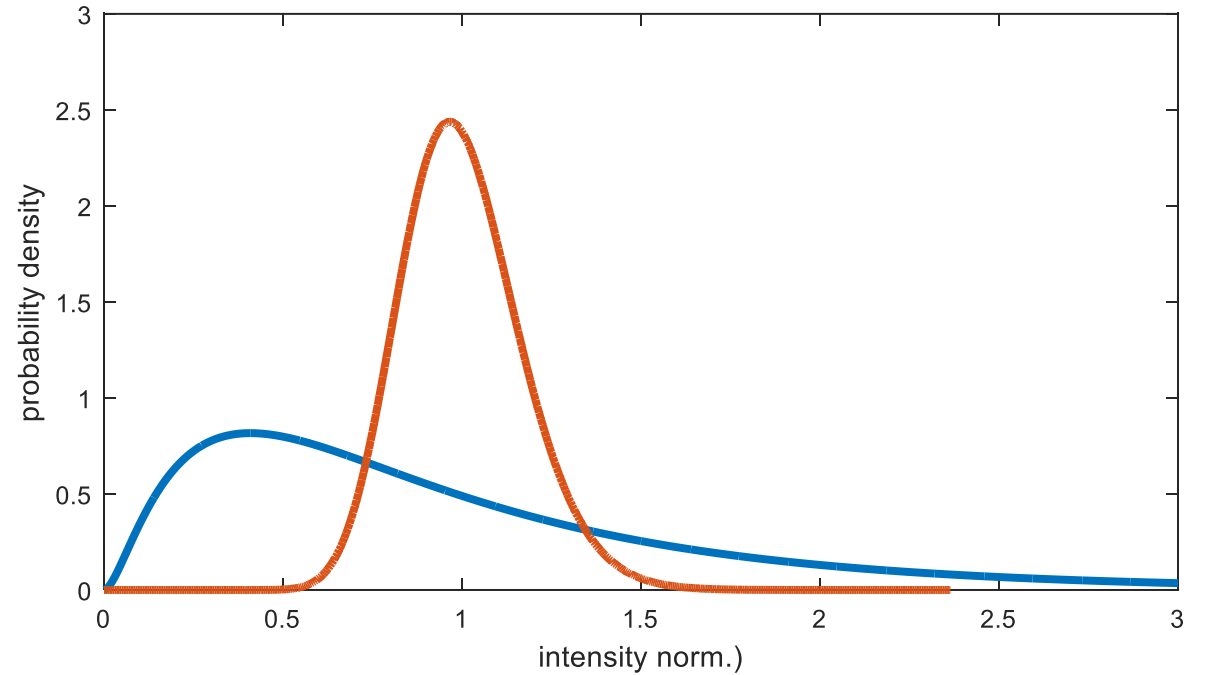
- Works for beam spreading, fiber coupling, limited (still) for intensit

4. Interleaving and Error Correction

- Can be very eff

5. Network orches

- Requires netwo



- Improve the curve ... Adaptive Optics, Spatial Diversity
- Deal with the curve ... Forward error correction, Interleaving, Adaptive transmission,
- Find a better curve (link) ... Network orchestration

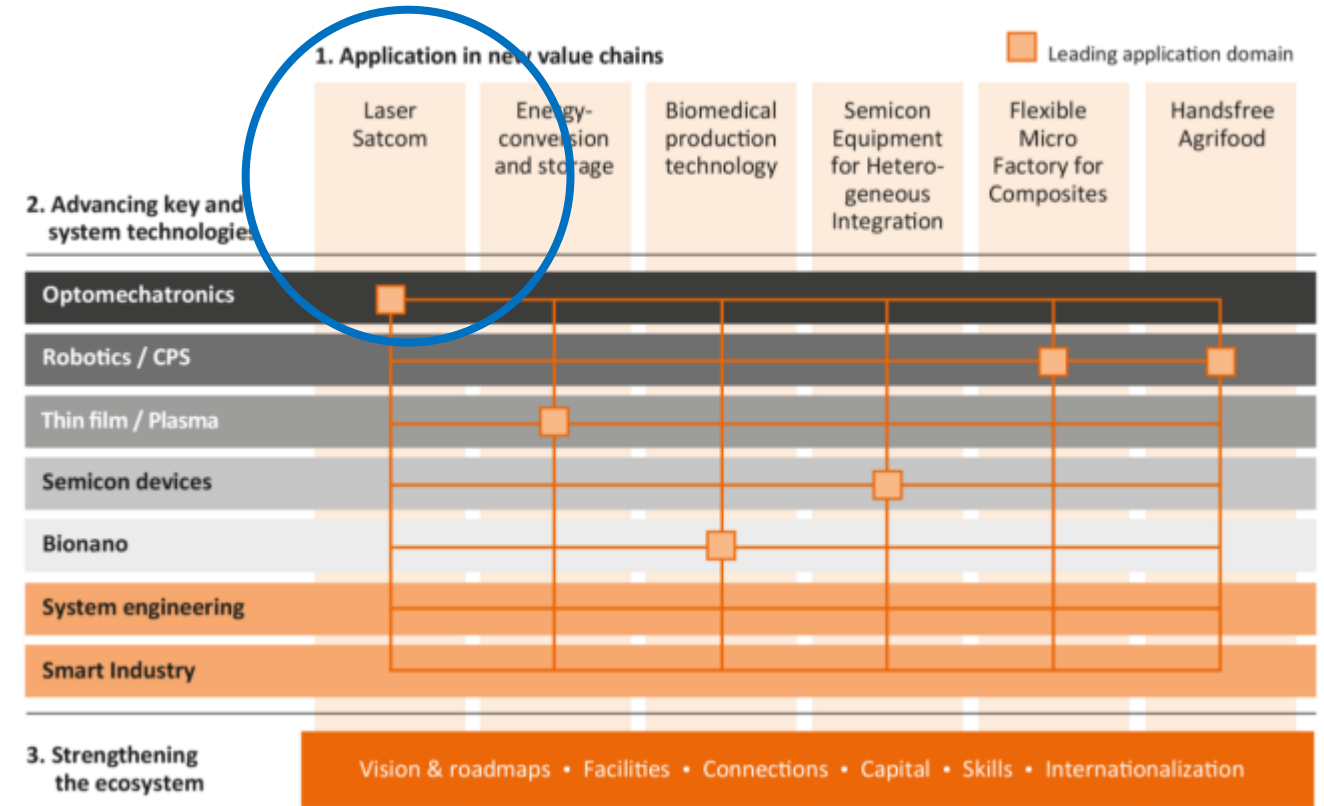
NATIONAL DEVELOPMENTS

GroEIFonds

- › NXTGEN HighTech proposal
- › Funded for 450 Meuro (April 2022)



The proposed solution: towards a coherent, leading high tech equipment ecosystem



Starting right away with over 340 companies and knowledge institutions

The parties participating in NXTGEN HIGHTECH consist of 189 SMEs (including 75 young start-ups and scale-ups), 130 larger companies and 23 knowledge institutions. Roughly 1500 researchers and innovators are contributing. With a letter of intent these organisations have shown their commitment

NL National research program – Optical Wireless Superhighways

TNO innovation
for life

- › Funded by NWO (Dutch Research Council)
- › Wireless links: from indoor links to satellite links
- › 5 universities, 15 PhD students
- › Period: 2022-2026
- › Active participation by industry (~20 parties)



› Research high-lights

- › Beam steering
- › Device localization
- › Diversity to reduce fading
- › Software defined networks
- › Multi-beam terminal
- › Key authentication
- › Low noise detection
- › Adaptive beam correction

TU Delft

TU/e

 **Universiteit
Leiden**

TNO

**UNIVERSITY
OF TWENTE.**

VU  **VRIJE
UNIVERSITEIT
AMSTERDAM**

opnt.
TIME PROVISIONING

AIRCISION

DEMCON | Focal

signify

phix

EFFECT
PROTONICS

QUIX
THE FLEXIBLE WAY TO MANUFACTURING

esa

LIONIX
INTERNATIONAL

FSO
INSTRUMENTS

AIRBUS
DEFENCE & SPACE

ISIS

ntr

SRON

SINGLE QUANTUM

OKO
OPTICAL

VTEC
SERVICES & SOLUTIONS

HYPERION TECHNOLOGIES

s & t

NXP

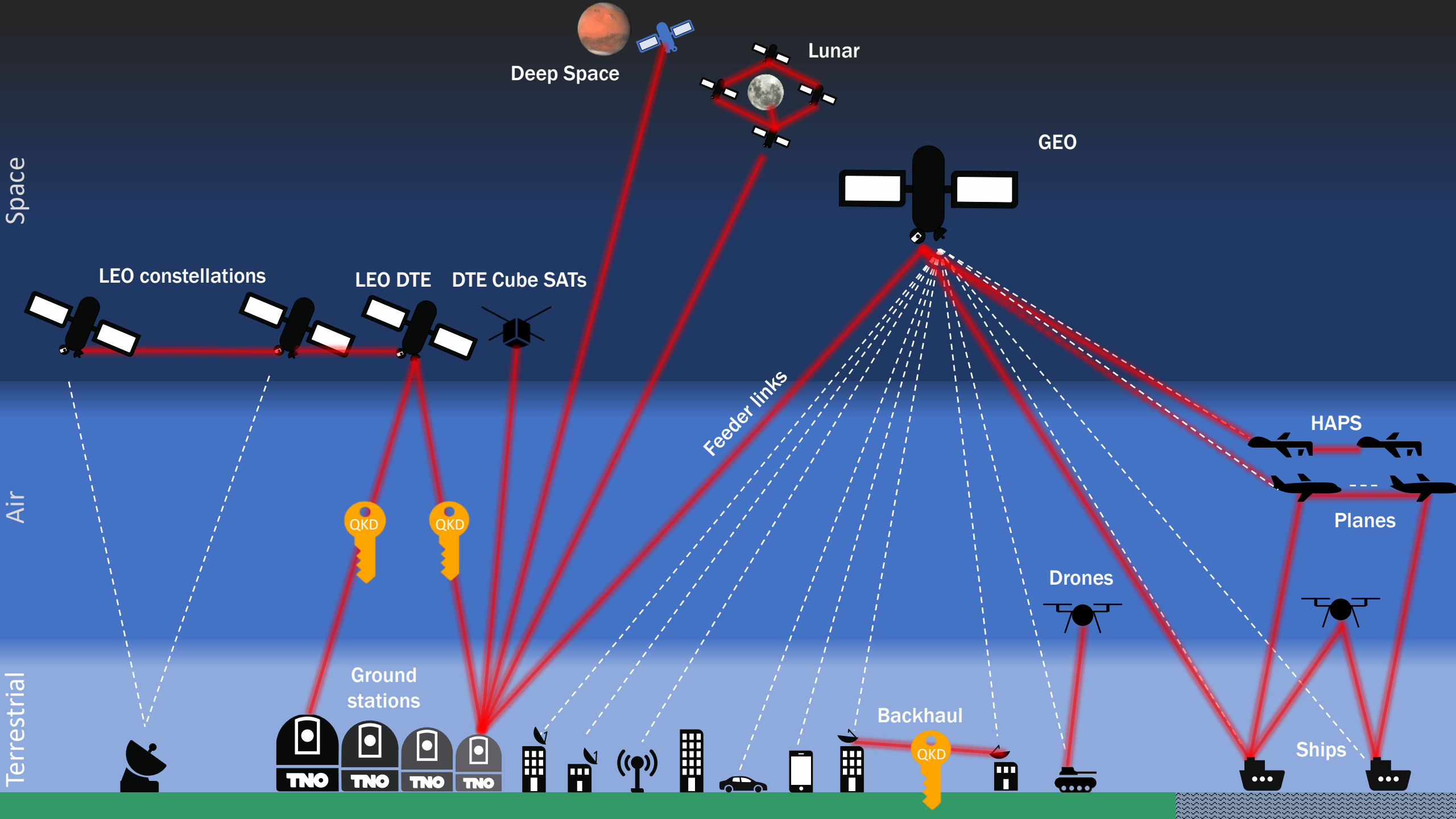
kpn

ABN-AMRO

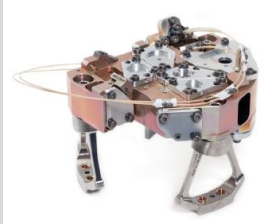
VDF

ASML

SUMIPRO
Optical solutions



Tesla Optical bench



LEOCAT optical head



CubeCAT



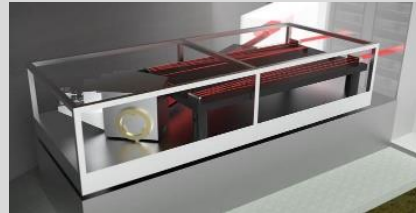
Fine steering mirror



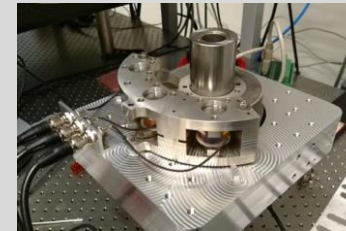
TOmCAT Tbit class terminal



Bulk multiplexer



Course pointing assembly



Pointing testbed



OC Lab



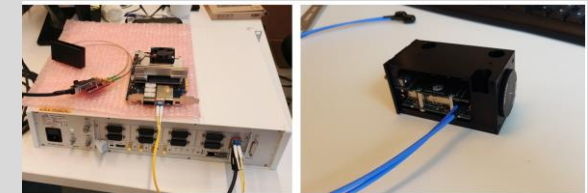
GoCAT Gbit class terminal



Deformable mirror



Modems/ detectors



FEEDER LINK GND TO GEO SATELLITE

› Terabit Optical Communication Active Terminal = TOmCAT

Partners:

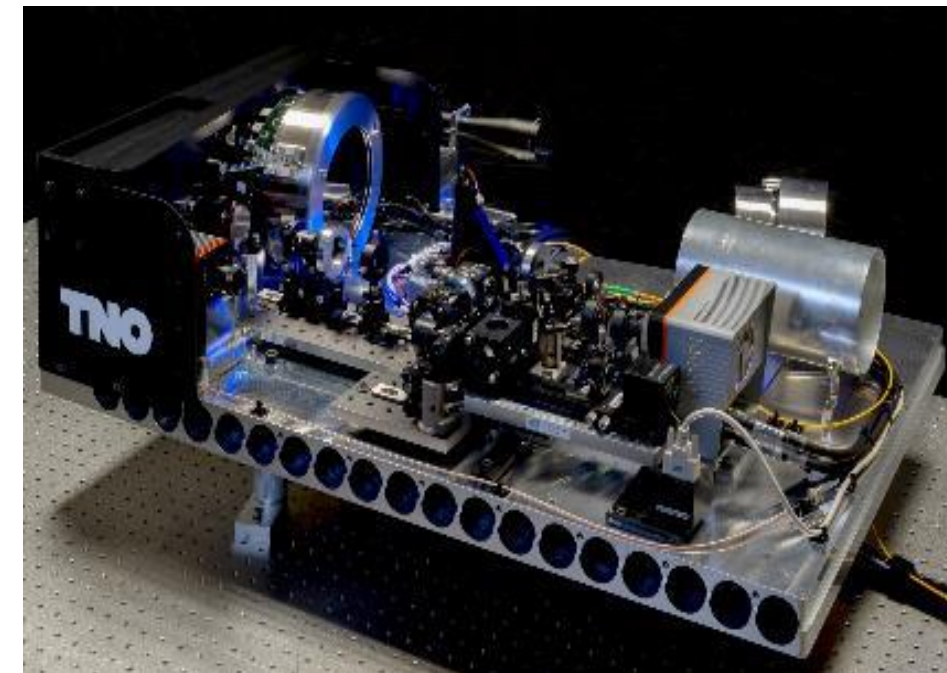
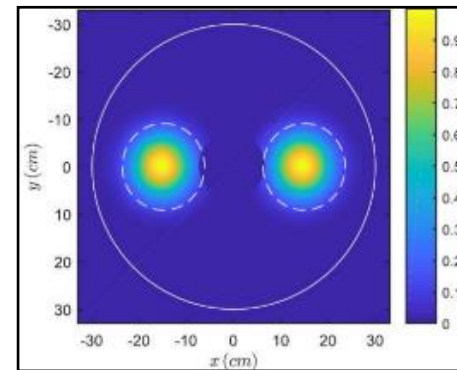


› Key aspects:

- › Adaptive Optics (AO) for pre-correction of atmospheric disturbances
- › Uplink, 13 channels x 100 Gbit/s, C-band
- › Downlink, 7 channels x 100 Gbit/s, L-band
- › 50 Watt per channel uplink
- › Bulk optics multiplexing

› Telescope:

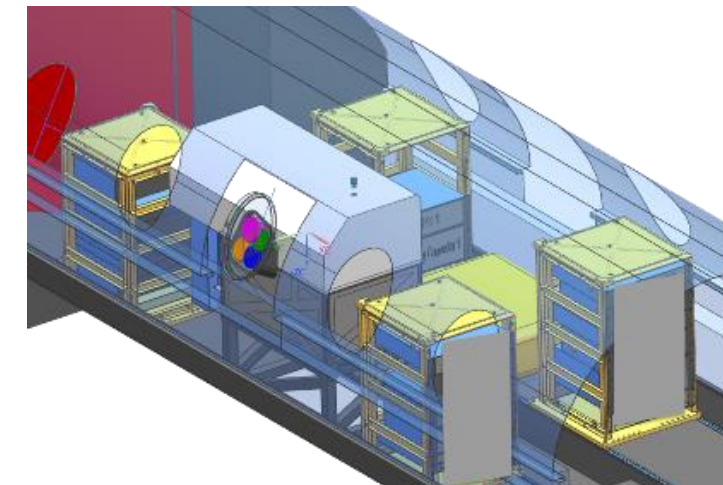
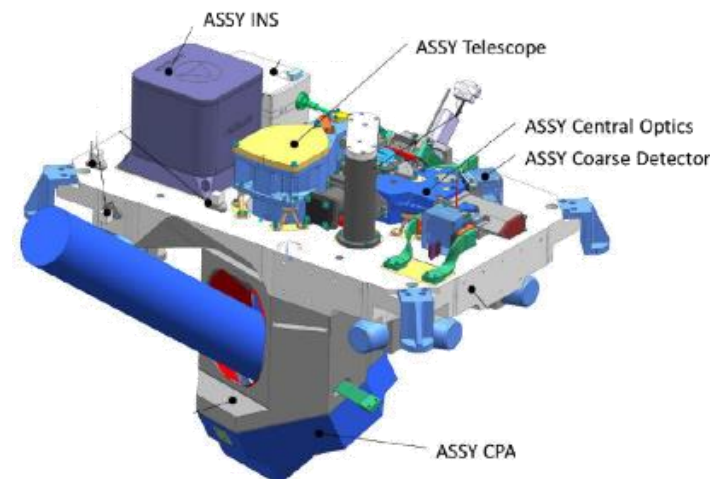
- › Rx aperture: 600 mm
- › Tx sub-aperture: 200 mm (2x)
- › Total transmitted power: 650 Watt



AIRBORNE LCT DEMONSTRATOR

- › Air based Laser Communication Terminal = Air-LCT
- › Compatible with European Data Relay System (EDRS)
- › Roadmap for full integration into aircraft
- › Current status:
 - › Demonstrator for optical communication through aircraft window
- › Pointing Challenge !
- › TNO Dynamical Optical Performance Evaluation = DOPE analysis
 - › Design based on end-to-end modelling, including:
 - › Optical modelling,
 - › FEM modelling,
 - › Dynamics
 - › Control modelling,
 - › Error sources

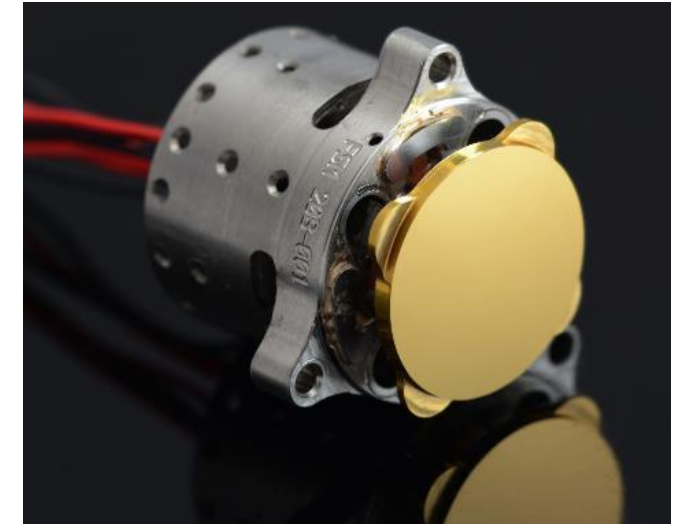
Dutch Industrial partner:



Dutch Industrial partner:



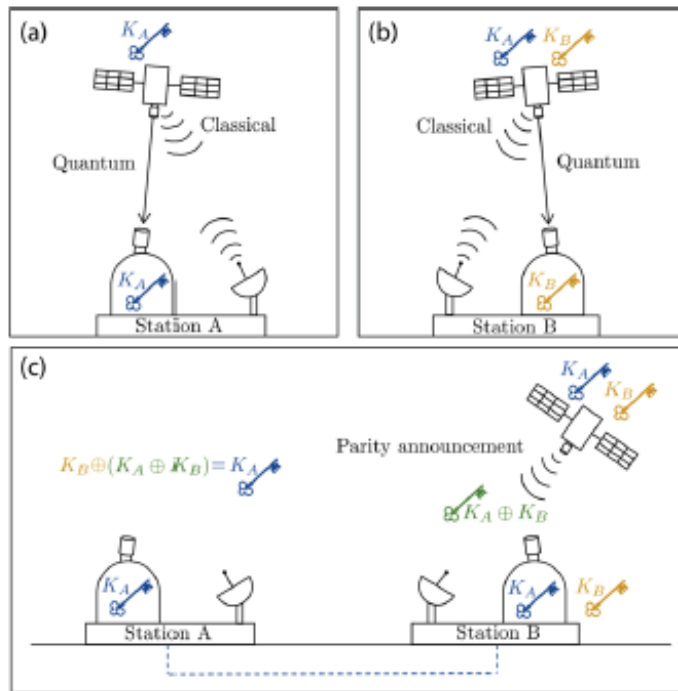
- › Fine/Fast Steering Mirror = FSM
- › TNO inhouse designed
 - › custom developed reluctance actuators, redundant coil windings
 - › hysteresis free flexural bearing
 - › mirror angle control via eddy current sensors
 - › dedicated thermal design enabling high optical power relay
- › Applications:
 - › CubeCAT terminal
 - › LEOCAT terminal
 - › Multi-purpose TNO breadboards for demonstration purposes
- › Space qualified:
 - › Vibration tested
 - › Thermal vacuum tested (*mechanical performance and outgassing*)
- › Status:
 - › Commercially available product through FSO Instruments



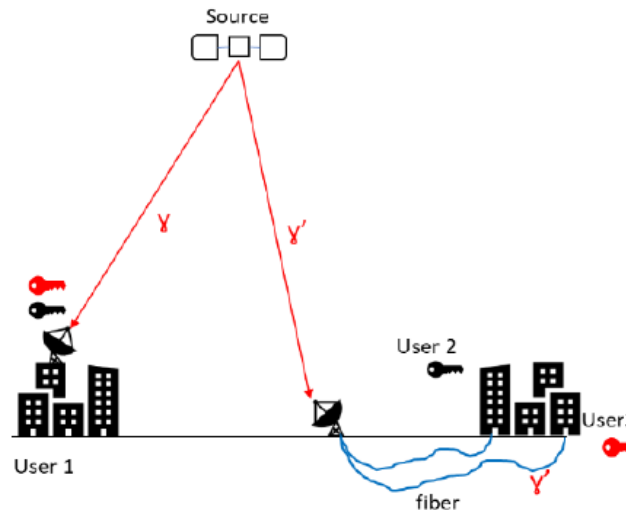
Tip/tilt range	$\pm 2^\circ$, i.e. $\pm 34,9$ mrad (optical)
Bandwidth (-3 dB)	> 1 kHz
Jitter	< 1 μ rad
Optical coating	Enhanced gold, >98% refl. @1550 nm
Max. optical power	~10 W
Mirror diameter	$\varnothing 20$ mm (flatness <12 nm rms) $\varnothing 30$ mm possible
Volume	$\varnothing 24$ mm x 30 mm

LONG RANGE KEY DISTRIBUTION SECURE COMMUNICATION WITH QKD

- Space/satellite based QKD
- Overcome distance limitation of fibers

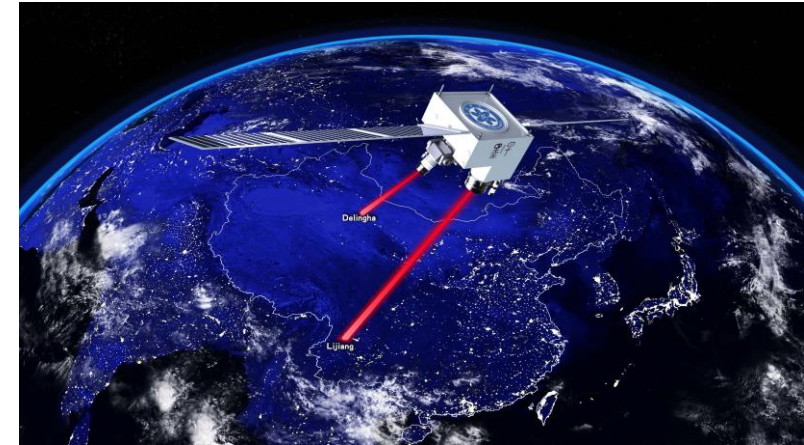


Bedington 2017

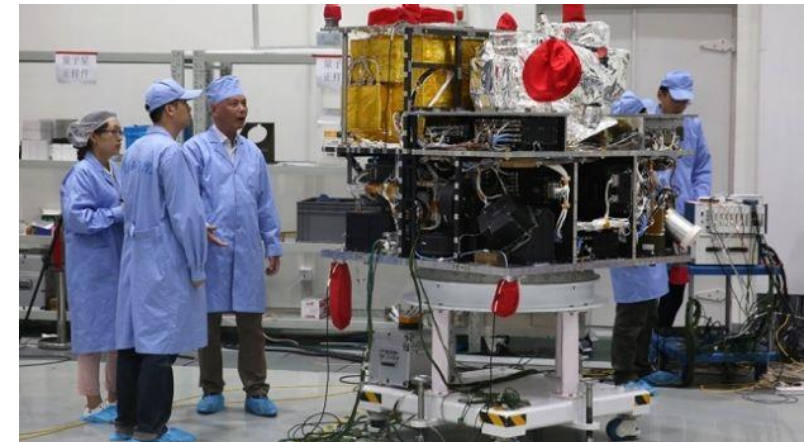


Courtesy Bob Dirks

Chinese quantum project QUESSE showed it worked!

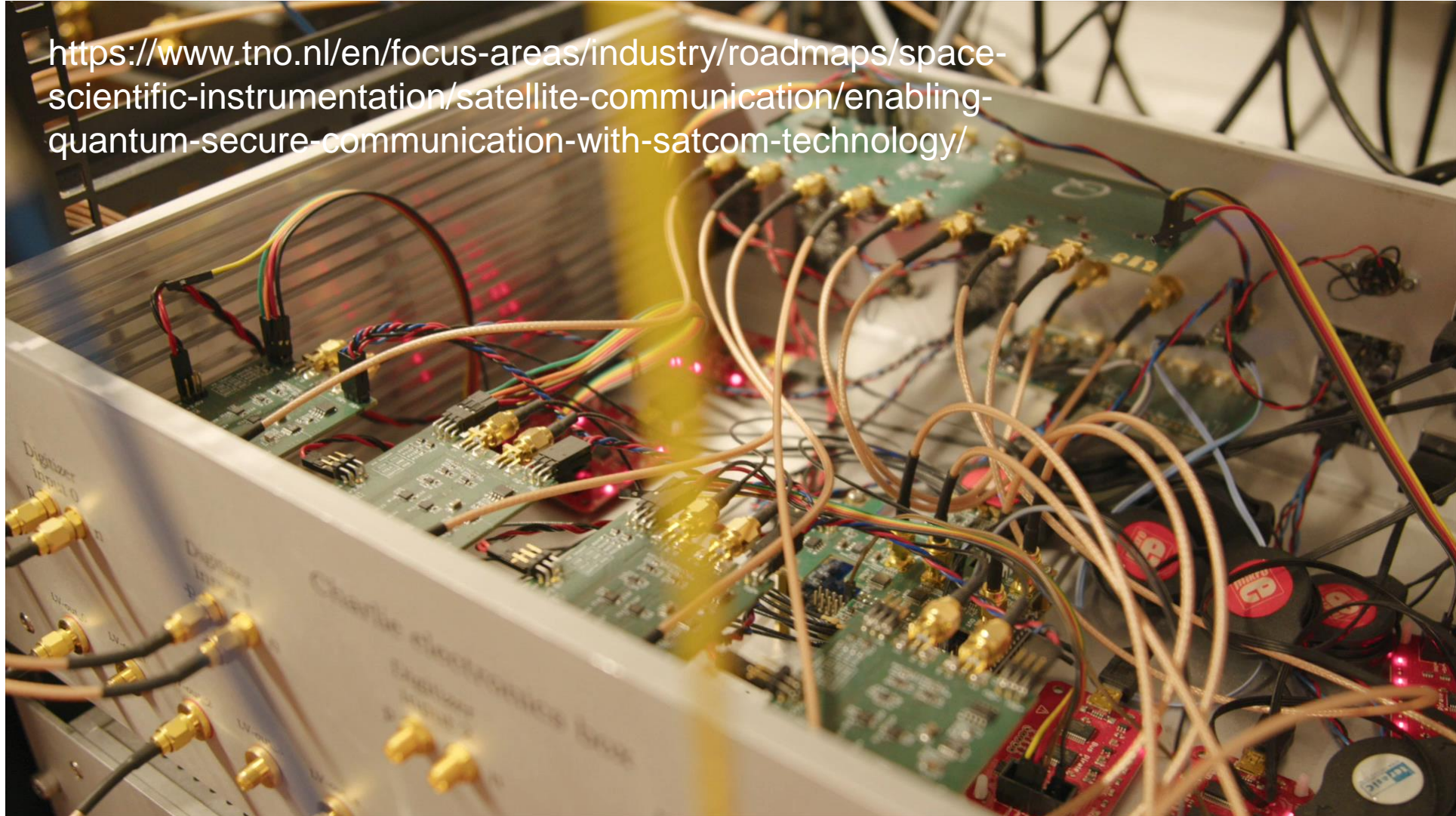


Micius satellite



QUANTUM KEY DISTRIBUTION

<https://www.tno.nl/en/focus-areas/industry/roadmaps/space-scientific-instrumentation/satellite-communication/enabling-quantum-secure-communication-with-satcom-technology/>





› **THANK YOU**

TNO innovation
for life